

Design of Electrical Systems for Rocket Propulsion Test Facilities at the John C. Stennis Space Center

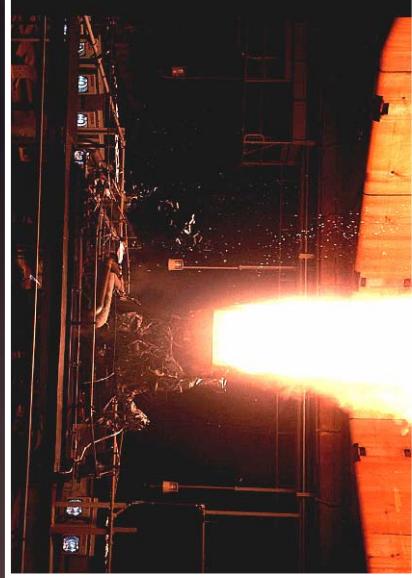


Stennis Space Center



RS-68 650 klbf
@ B1 Test Stand

Space Shuttle Main Engine Test
@ A2 Test Stand



Fastrac 60 klbf
@ B2 Test Stand

Design of Electrical Systems for Rocket Propulsion Test Facilities at the John C. Stennis Space Center



Stennis Space Center

- ◆ NASA/SSC's Mission in Rocket Propulsion Testing Is to Acquire Test Performance Data for Verification, Validation and Qualification of Propulsion Systems Hardware

- Accurate
- Reliable
- Comprehensive
- Timely

- ◆ Data Acquisition in a Rocket Propulsion Test Environment Is Challenging

- Severe Temporal Transient Dynamic Environments
- Large Thermal Gradients
- Vacuum to 15k psi pressure regimes

- ◆ SSC Has Developed and Employs DAS, Control Systems and Robust Instrumentation that Effectively Satisfies these Challenges

- ◆ The Following Presentation Reviews SSC's Data Acquisition and Controls Architectures



Agenda

- ◆ **Background – SSC EE Org & Test Facilities**
- ◆ **High/Low Speed Data Acquisition Systems**
- ◆ **Control Systems**
- ◆ **Data Acquisition and Control Systems Lab**
- ◆ **Unique Sensor Development Activities**

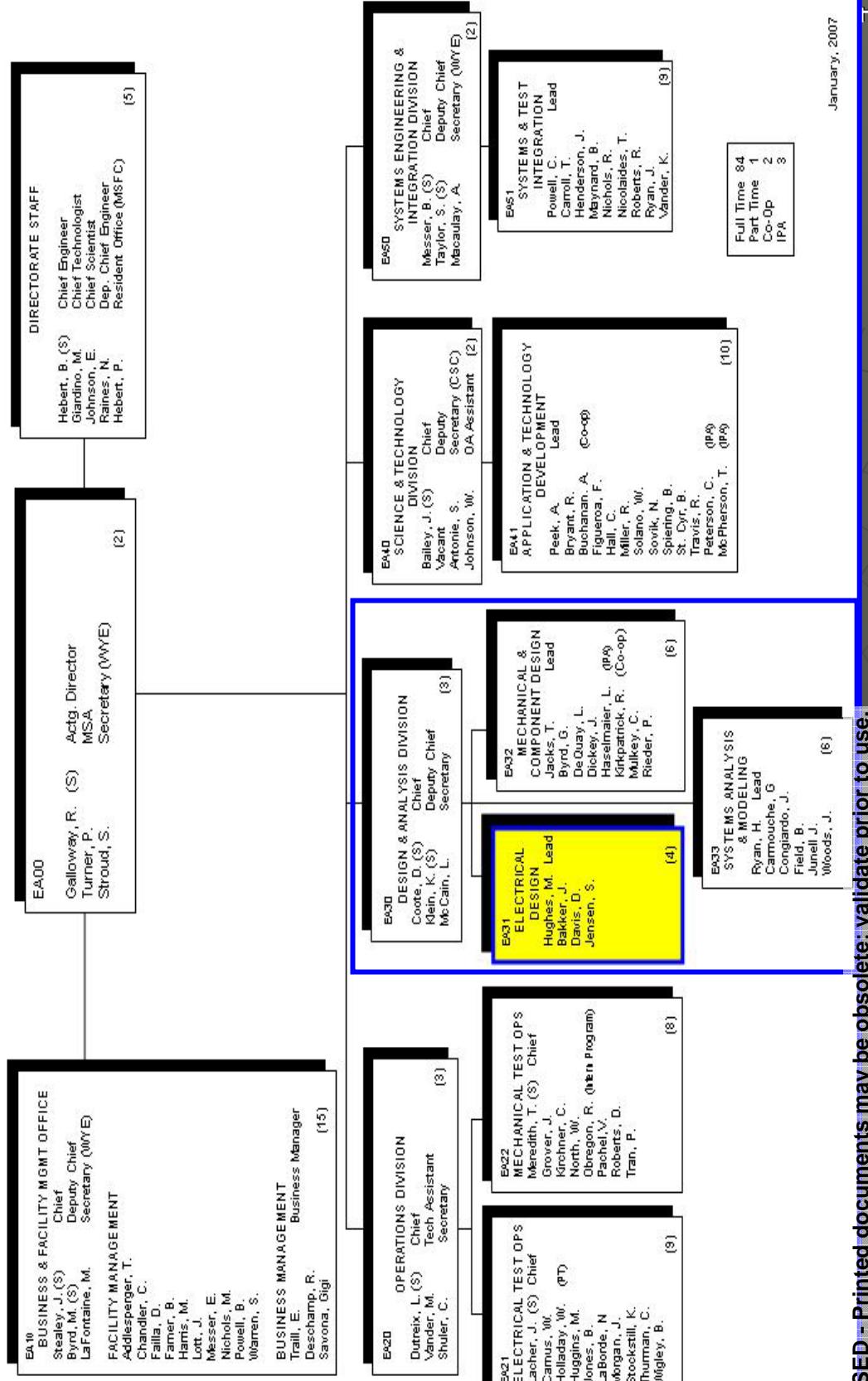




ENGINEERING & SCIENCE DIRECTORATE

Stennis Space Center

ENGINEERING & SCIENCE DIRECTORATE



Design & Analysis Division



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Design and Analysis Division

- Configuration Management
- Records Retention DB Management

Mechanical and Component Systems

- Cryogenic Propellant Systems
- Storable Propellant Systems & HPIW
- Hydraulics/pneumatics Systems
- Press Gas/Purge Systems (TBA)
- Components
- Materials
- Ancillary Systems
- TMS, Measurement Uncertainty
- Standards & Specifications

Electrical Systems & Software

- Data Acquisition
- Instrumentation & Signal Conditioning
- Controls & Simulation
- DACS Lab Management
- Data Systems Management
- Ancillary Systems/Electrical Power

Systems Analysis & Modeling

- Modeling and Analysis development and integration into RPT
- Fluid Mechanics/Thermal Analysis of Propellant Systems
 - Liquid
 - Gas
- CFD
- Structures/Loads Analysis
- Thermal/Heat Transfer Analysis

Organization Goal:

- Develop and maintain propulsion test systems and facilities engineering competencies
 - Unique and focused technical knowledge across respective engineering disciplines applied to rocket propulsion testing. e.g.,
 - Materials selection and associated database management
 - Piping, electrical and data acquisition systems design for cryogenic, high flow, high pressure propellant supply regimes
 - Associated analytic modeling and systems analysis disciplines and techniques
- Printed documents may be obsolete: validate prior to use.



SSC Test Facilities



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AB-Complex

A-1 Full Scale Engine Devt. & Cert
J-2X **A-2** **SSME**



Components
...Engines
... Stages

B-1/B-2
Full Scale Engine/Stage
Devt. & Cert
RS-68/ARES



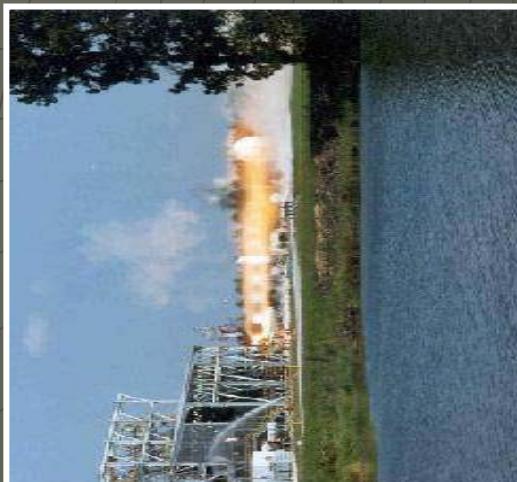


E-Complex

E-1 Cells 1, 2, 3

High Press., Full Scale
Engine Components

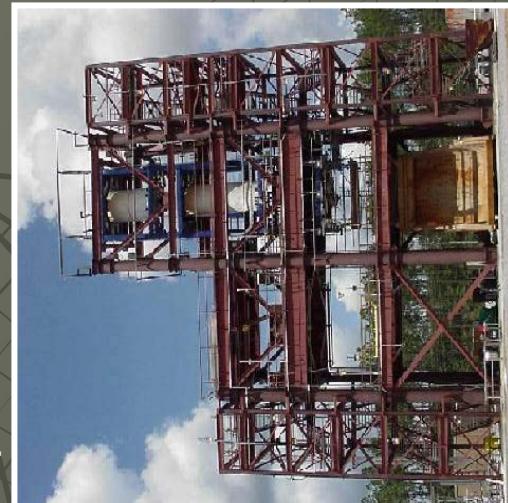
J-2X



E-2 Cell 1

High Press.
Mid-Scale
& Subscale

J-2X



E-2 Cell 2 Low Press. Mid-Scale & Subscale, Stage

High Press. Small-Scale
Subscale

E-3

Obsolete; validate prior to use.



TGV

E-3

Cell 2

Obsolete; validate prior to use.

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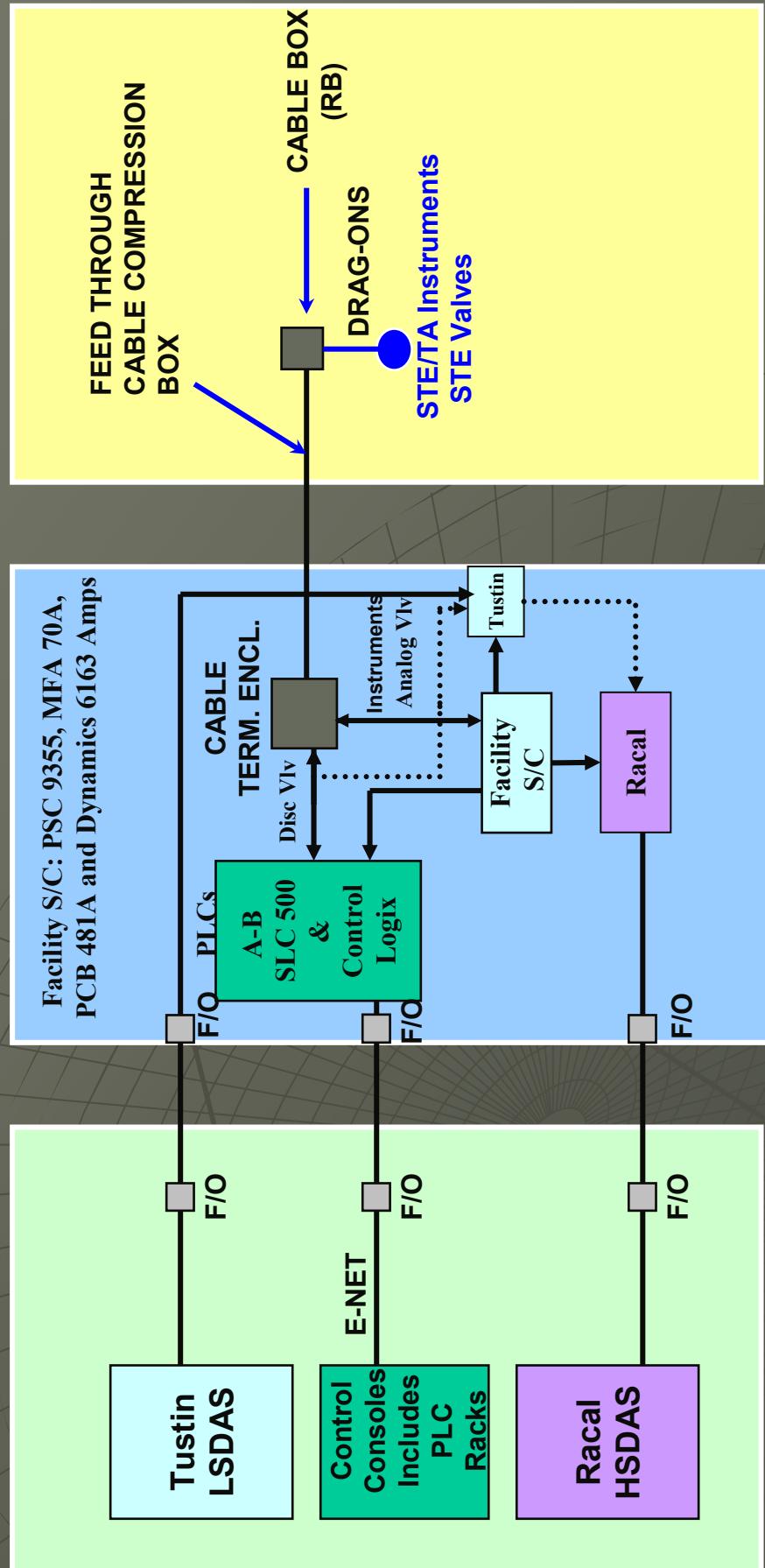
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Test Facility Layout

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Test Article

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Test Control Centers

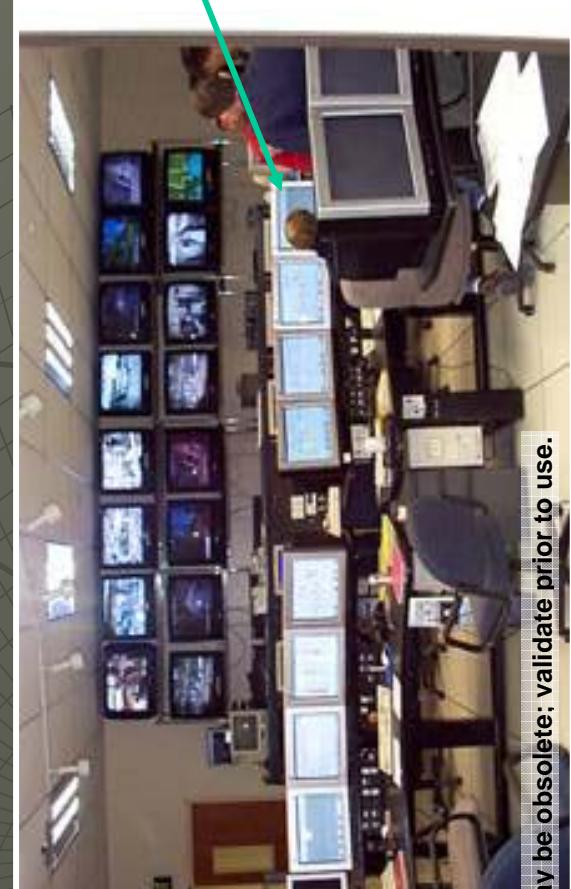
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A2 TCC



Test Conductor's Station



E2 TCC





Signal Conditioning Buildings (SCB)

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E1 SCB
Signal
Conditioning
Rack



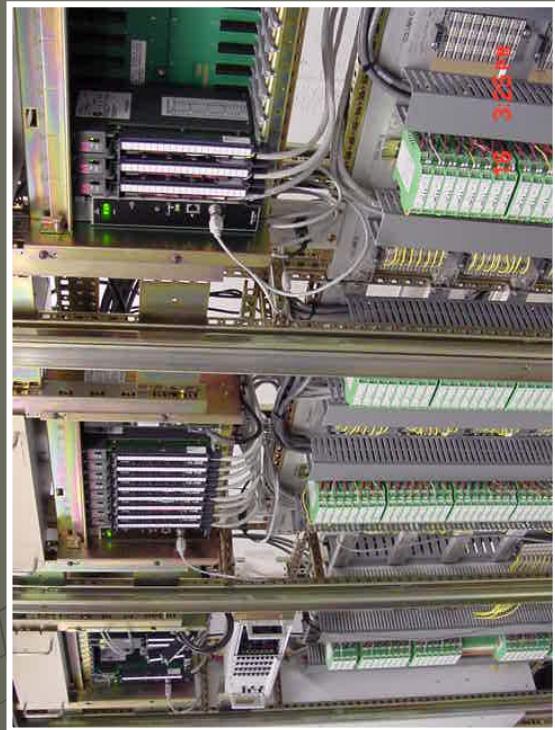
E2 Cell 1
SCB 1
Controls
Racks



E2
SCB's
1 & 2

16 3:16 PM

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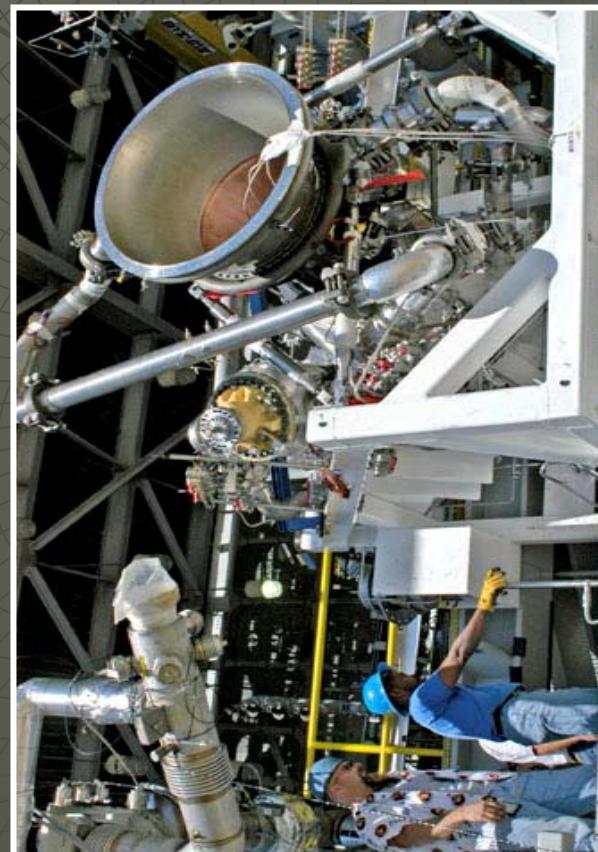


Typical Test Articles

Stennis Space Center



LR-89



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**Integrated Powerhead
Demonstrator**



Test Facility Electrical Systems

- ◆ Communications System
- ◆ **Control System**
- ◆ Facility Fire Alarm System
- ◆ Fire & Gas Leak Detect System
- ◆ Grounding System
- ◆ **High Speed Data Acquisition System**
- ◆ Lighting System
- ◆ Lightning Protection System
- ◆ **Low Speed Data Acquisition System**
- ◆ Oral Warning System
- ◆ Power Distribution System
- ◆ Uninterruptible Power System
- ◆ Video System



High Speed Data Acquisition Systems (HSDAs)

Mark Hughes





High Speed Data Acquisition System

Stennis Space Center

- The High Speed Data Acquisition System is used to record rocket engine or component data from a variety of dynamic sensors.
- Sampling rates normally range from 5.12K to 204.8K samples per second (For Comparison, the Low Speed Data Acquisition System ranges from 1 to 250 samples per second.)
- High speed data provides the Analyst with information about the dynamic environment/condition of a test article. The data feeds models that characterize the performance of the test article or allows the analyst to help determine the health of the hardware.
- Challenges to recording good high speed data include the environment (high temperatures, vibration, high flow, cryogenic temperatures, high pressure), proper cabling, appropriate sensor selection, and numerous other considerations.



High Speed

Data Acquisition Systems

Stennis Space Center

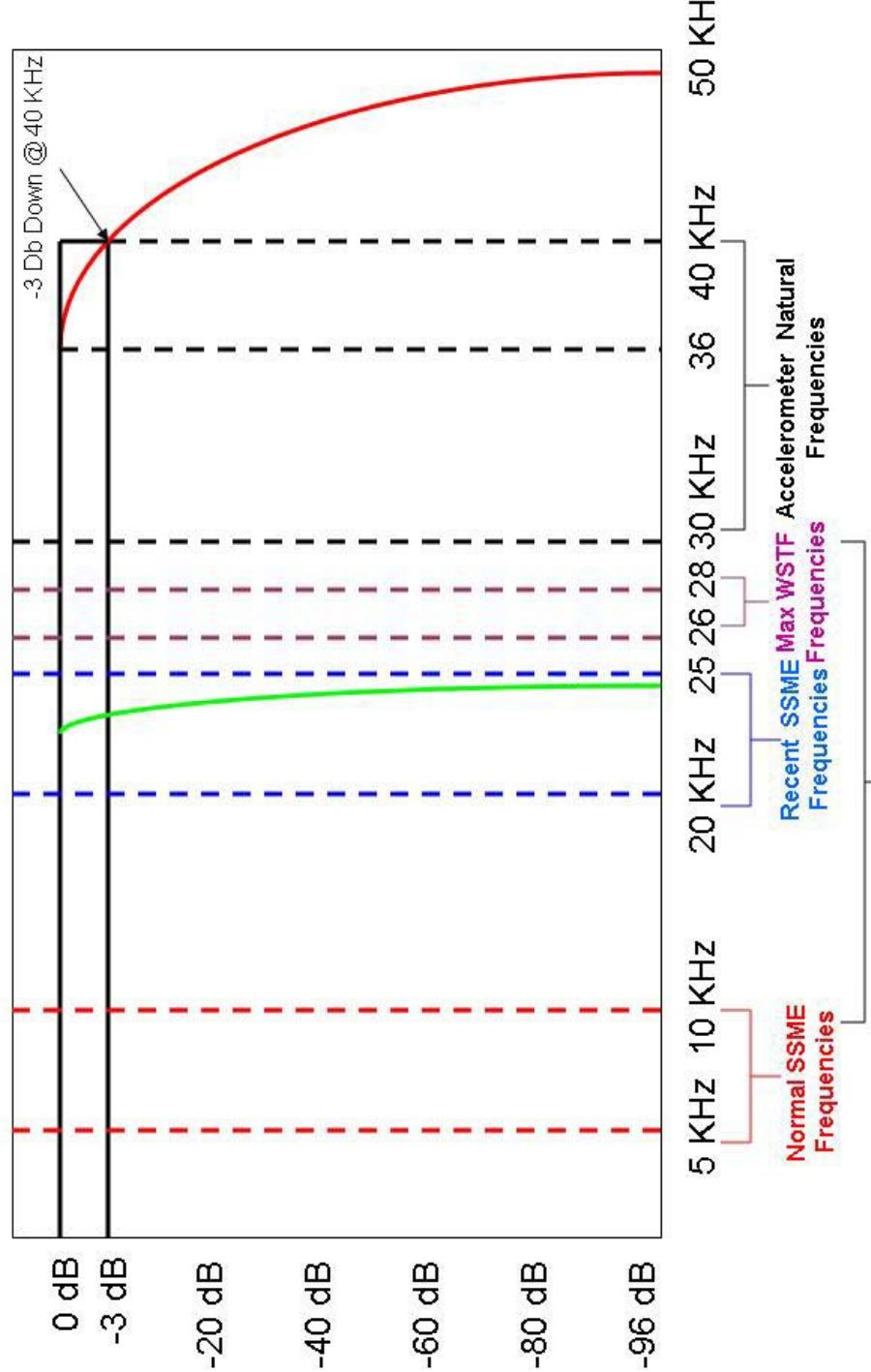
- ◆ **RACAL (Obsolete) - 100,000 Samples Per Second (**Decimal Sampling**)**
 - AB-Complex (SSME)
 - E-Complex (TGV)
- ◆ **MIDDAS (SSME Only)- 51,200 Samples Per Second (**Binary Sampling**)**
 - A Complex (SSME Only)
- ◆ **DataMAX II (New)- 204,800 Samples Per Second (**Binary & Decimal Sampling**)**
 - AB Complex (RS-68, J-2X)
 - E Complex



Typical HSDAS Bandwidth Usage



MIDDAS — HSDAS BANDWIDTH USAGE (102.4K SPS)
DataMAX —





RACAL HSDAS

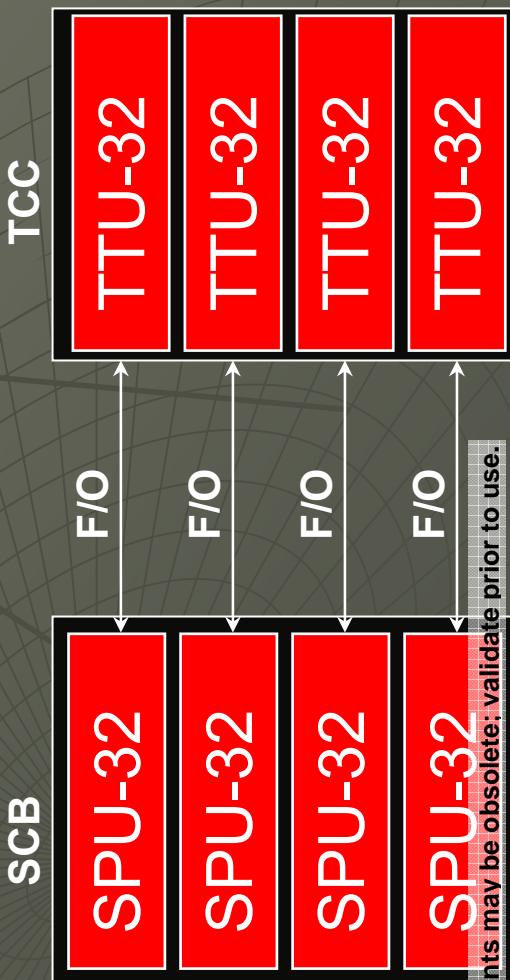
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- **Obsolete-Being Replaced by DataMAX II HSDAS**

Specifications

- 128 Channels
- 100K Samples Per Second
- 45.5 KHz Bandwidth
- 16 Bit Delta-Sigma A/D Conversion
- 0.5, 1, 2, 5, 10, 20 50 Volts Peak

- Decimal Sampling Only
- AB-Complex Use Ends in 2007, E-Complex by 2010



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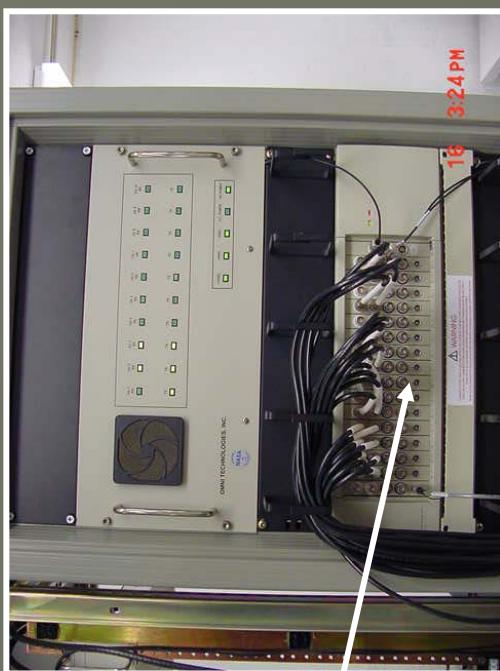
RACAL HS DAS

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Tape
Transport Unit

Signal
Processing Unit



Handheld
Controller

Direct-to-Disk
System



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MIDDAS HSDAS

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- SSME Use Only

- Binary Sampling

- Backed up by a DataMAX II

- Used for Quick Turnaround Data

TCC

VXI-16

No SCB
Hardware

Results in Cable Lengths
of around 1700 FT





MIDDAS HSDAS

Stennis Space Center

MIDDAS in Standalone Configuration



128 Channel MIDDAS System

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DataMAX II HSDAS

Stennis Space Center

- Mirrored Data Recording

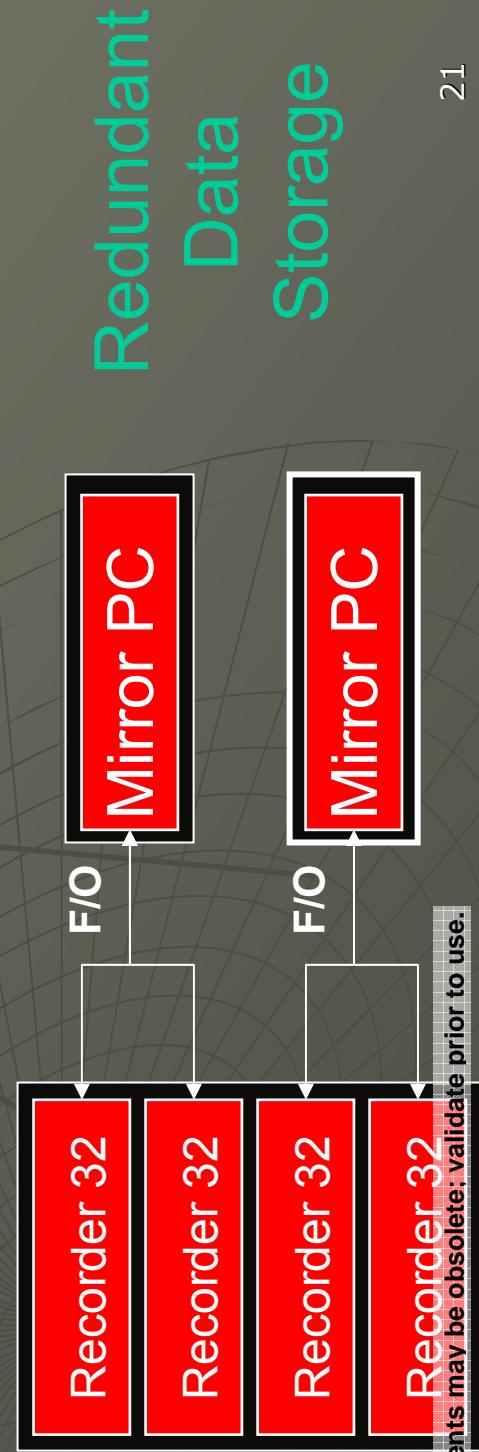
- Binary & Decimal Sampling

- Rates up to 204.8 K SPS

- Fast Turnaround and Archive Data

Specifications

- 192 Channels
- 204.8K Samples Per Second
- 90 KHz Bandwidth
- 16 Bit Delta-Sigma A/D Conversion
- 1, 4, 10, 40 Volts Peak

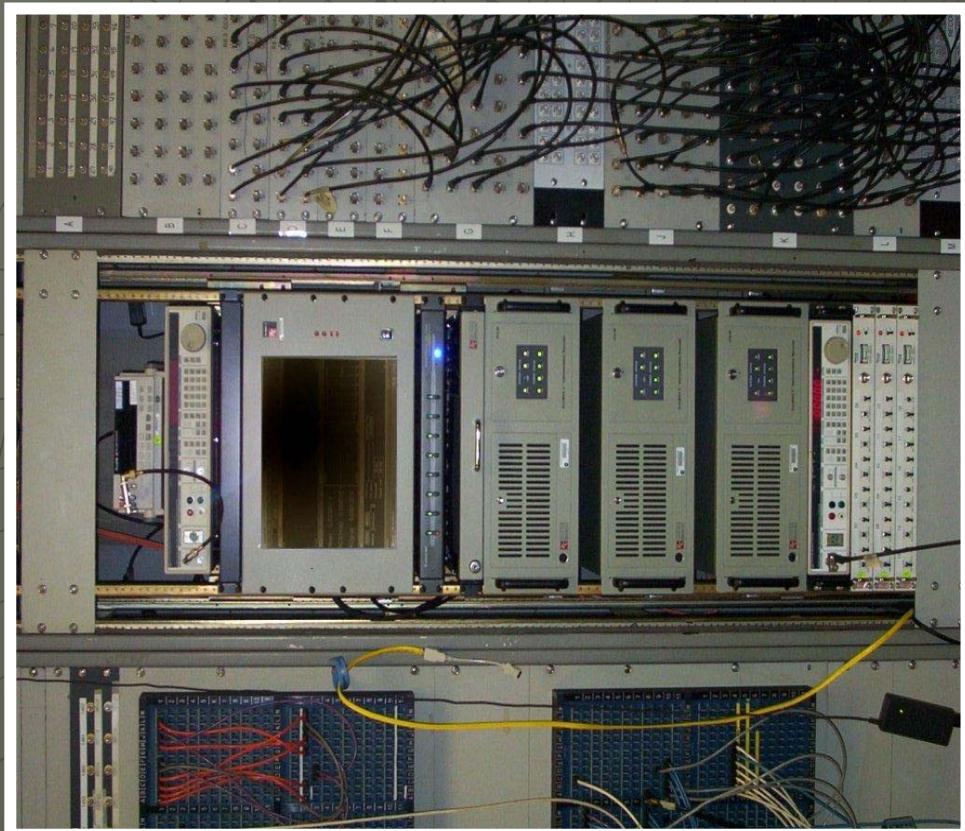




DataMAX II HSDAS

Stennis Space Center

Mirrored Drives in the TCC



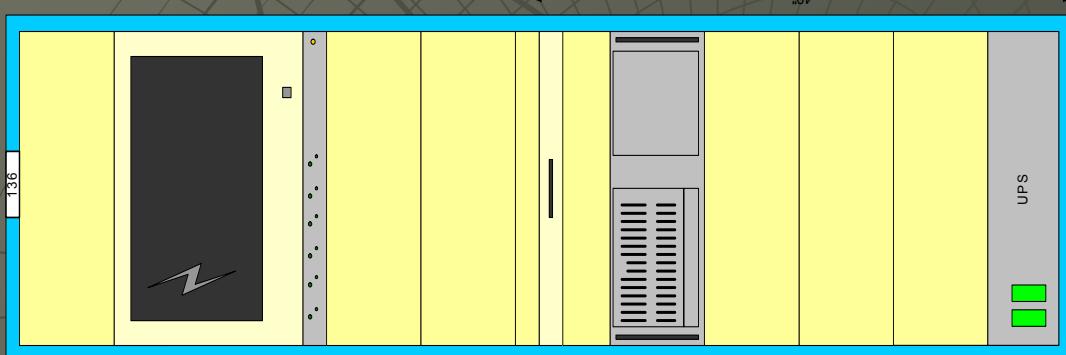
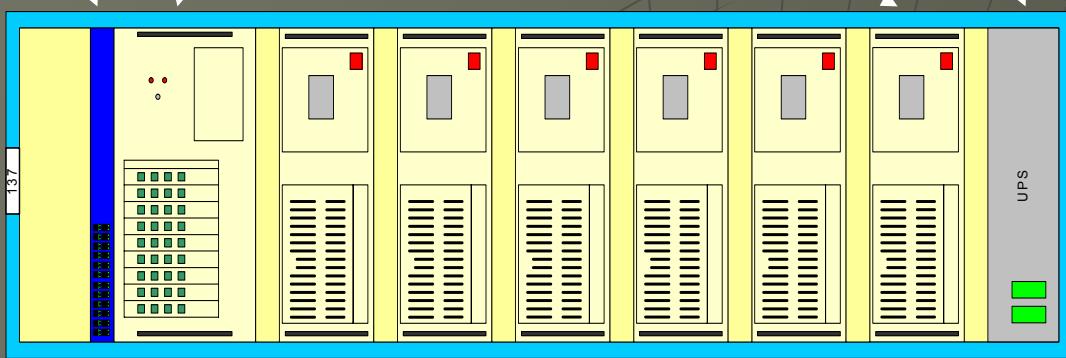
32 Channel Recorders

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DataMAX III

Planned for B2 Test Complex

Stennis Space Center

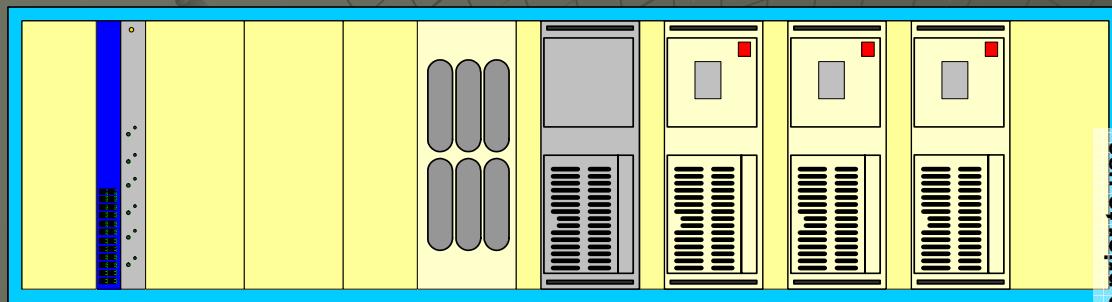


DataMAX II

Planned for B2 Test Complex

Stennis Space Center

Gigabit
Gear



KVM

USB
HDD's

Rack PC

**Test Control
Center**



Typical High Speed Data Acquisition System Instrumentation

Stennis Space Center

FACILITY

SPECIAL TEST EQUIPMENT

TEST ARTICLE

Typical Instrumentation
not always in the Catalog

- Special Ranges
- Temp Compensation
- Special Materials

Accelerometer
Strain

Dynamic Pressure
Accelerometer
Strain
Proximity
Speed

Typical High Speed DAS Instrumentation

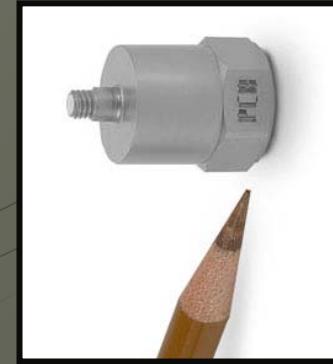
Typical High Speed Data Acquisition System Instrumentation



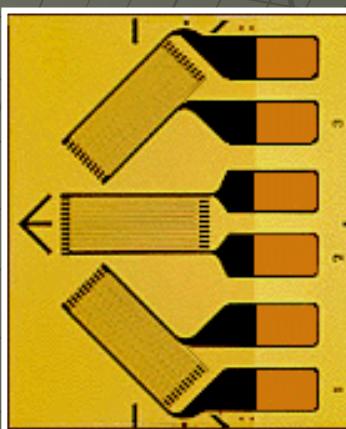
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Dynamic Pressure



Strain

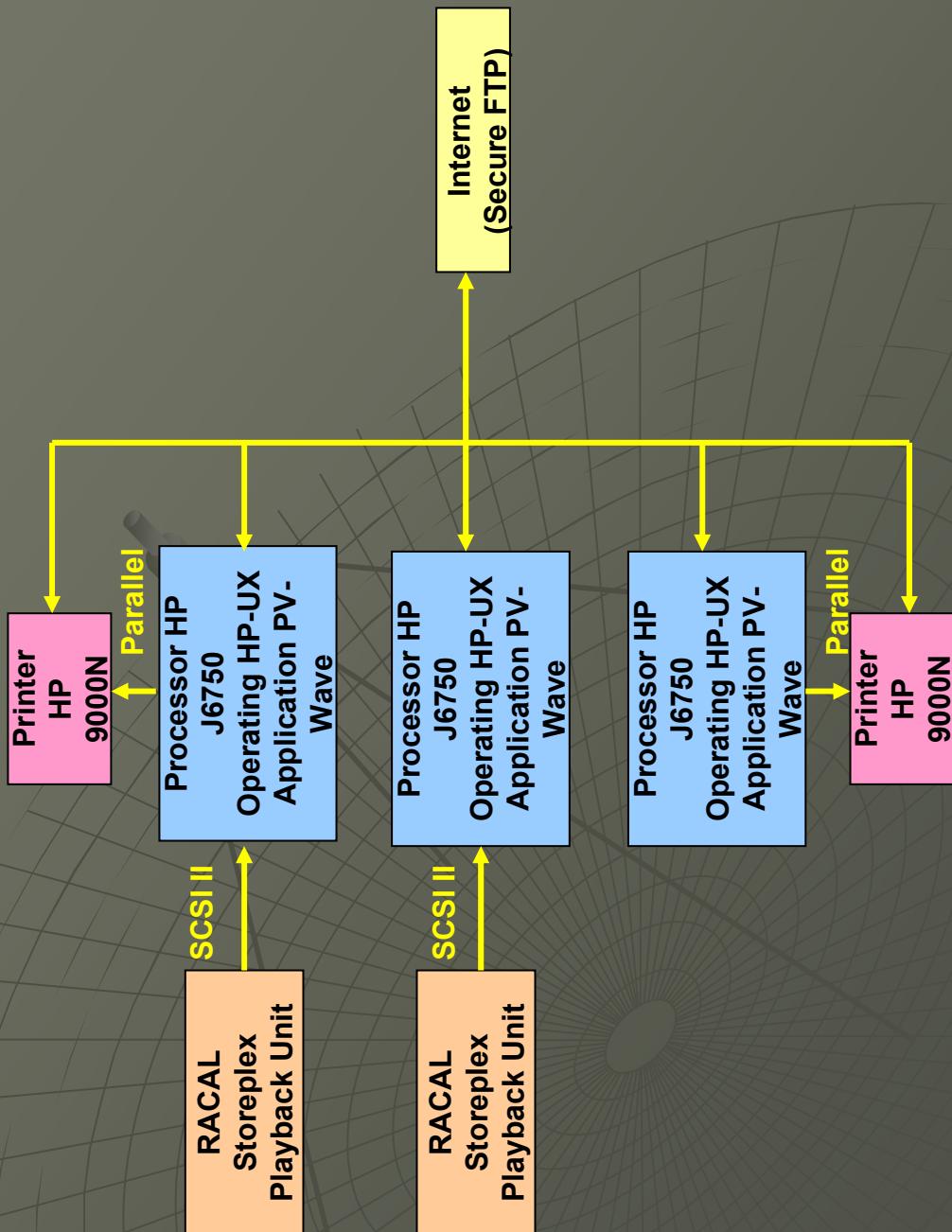


Accelerometer

Speed

E-Complex High Speed Data Processing System

Stennis Space Center

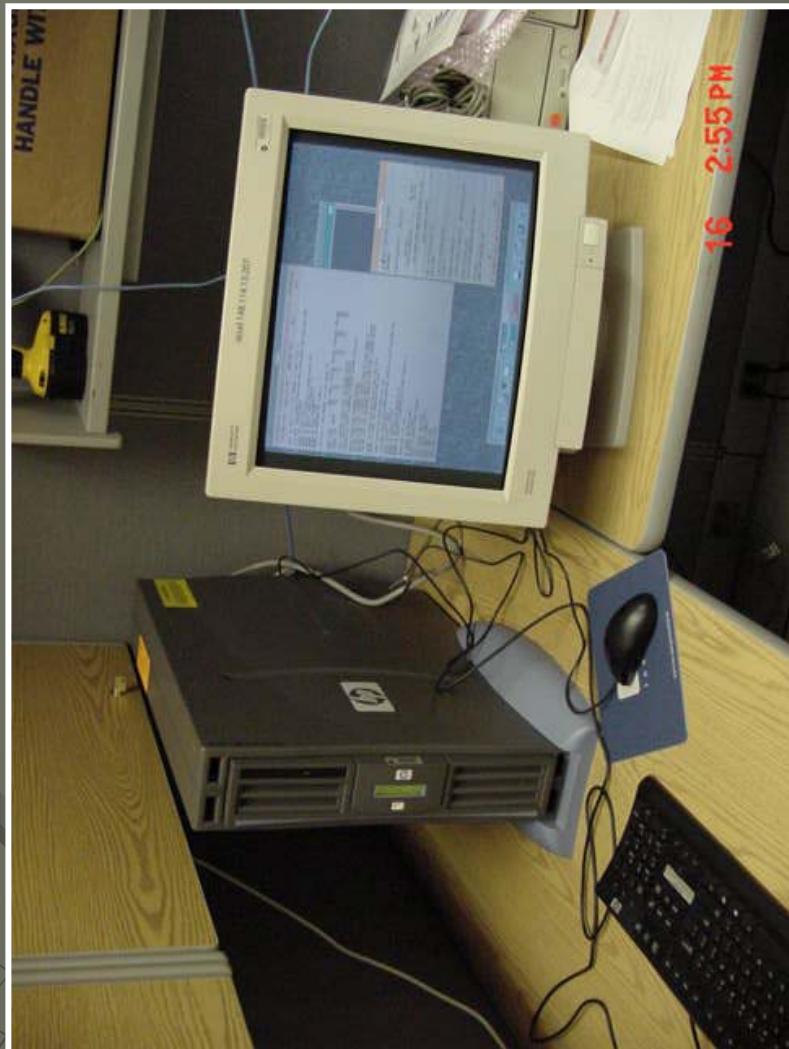




E-Complex High Speed Data Processing System

Stennis Space Center

HP J6750 Unix Workstations



- Twin 875 MHz RISC Processors
- 4 GB RAM
- 72 GB Storage





Low Speed Data Acquisition Systems

Dawn Davis





SSC's Low Speed

Data Acquisition Systems

Stennis Space Center

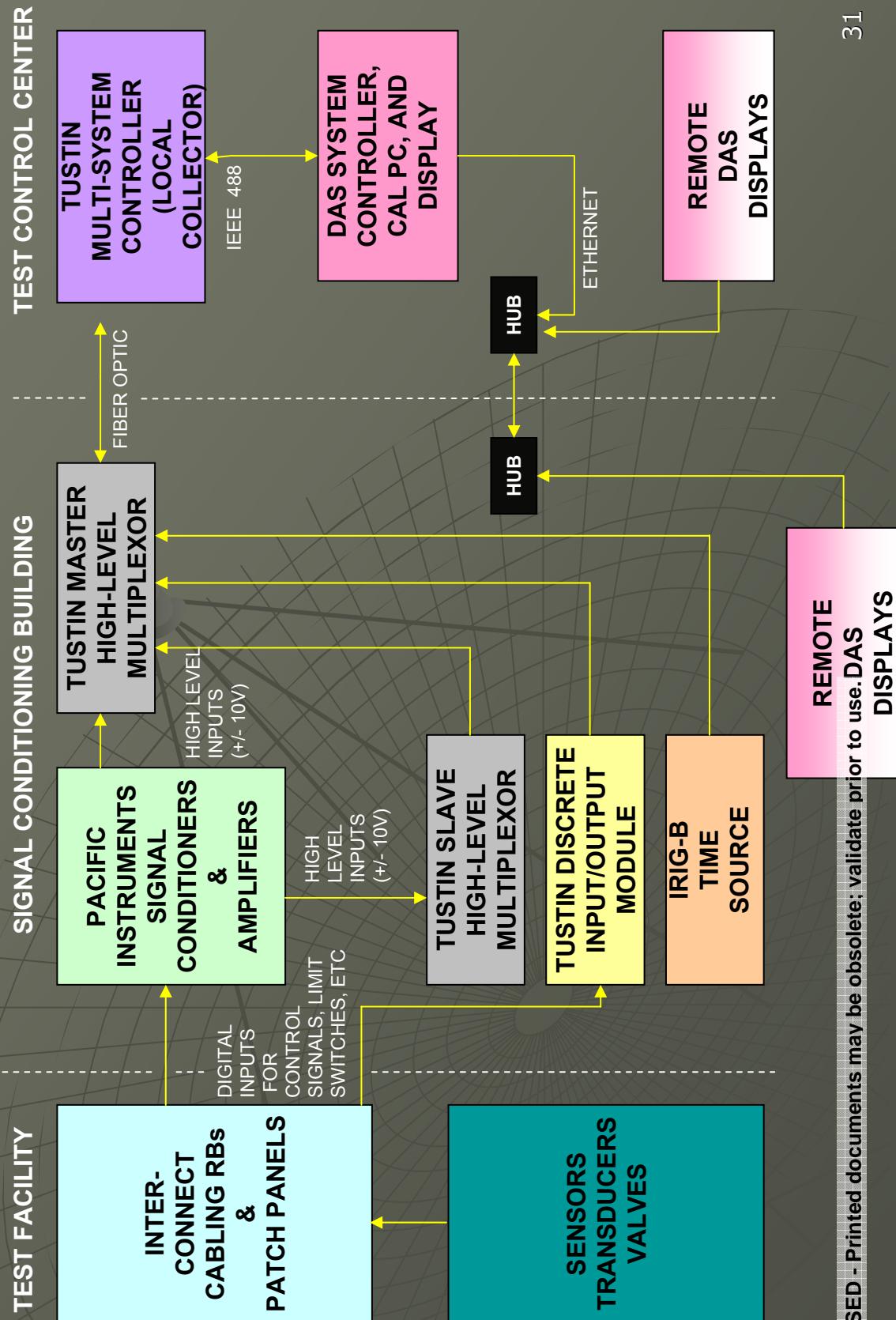
- ◆ **Data acquisition, recording, real time display**
- **Data types:** Low frequency Analog Data, Discrete (event) Data, Pulse Data from flow meters and speed sensors
- ◆ **E-Complex Tustin** - 250 samples per second
- ◆ **AB-Complex PreSys** 1000 - 250 samples per second



E-Complex Low Speed Data Acquisition System Architecture



Stennis Space Center





E-Complex Low Speed Data Acquisition System Architecture

Stennis Space Center

The E-Complex consists of three test stands

- **E1**
 - ◆ Contains four separate data systems: facility, Cell 1, Cell 2, Cell 3
 - ◆ Each system contains 512 analog input channels and 320 discrete channels
 - ◆ Two systems run during a test: Facility and cell
- **E2**
 - ◆ Contains two separate data systems: Cell 1 and Cell 2
 - ◆ Each system contains 400 analog input channels and 420 discrete channels
 - ◆ Systems include both facility and test cell measurements
- **E3**
 - ◆ Contains one data systems for both cells
 - ◆ System contains 400 analog input channels and 312 discrete channels



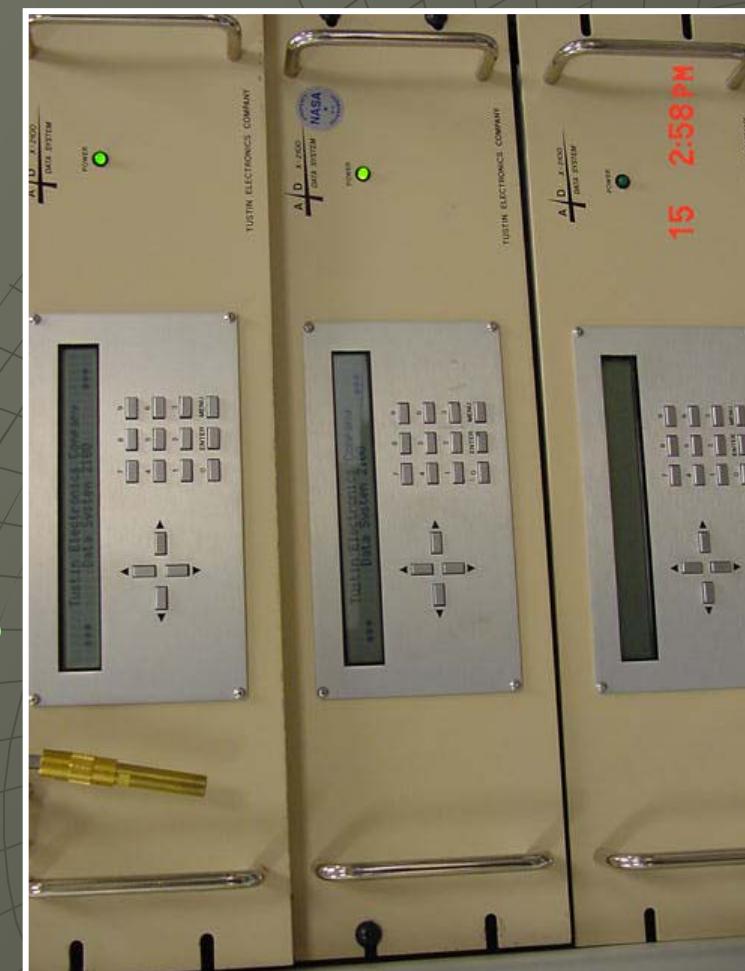


E-Complex

Low Speed Data Acquisition System

Tustin Data System

Stennis Space Center



- ◆ Fully populated analog box
 - 128 analog input channels

- ◆ Fully populated discrete box
 - 320 digital input channels

E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Pacific Instruments Signal Conditioners

Model 9355



- **Programmable**
 - ◆ Gain, filter, excitation
- **Automated Calibration**
 - ◆ Voltage Insertion
 - ◆ Shunt
 - ◆ Rcal
- **Various Completion Cards**
 - ◆ Full Bridge, Half Bridge
 - ◆ Internal or external shunt resistors
 - ◆ ICP
- **Measurements**
 - ◆ RTD's
 - ◆ Pressure Transducers
 - ◆ Strain Gauges

E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Pacific Instruments Amplifiers

Model 70A

- **Manual Settings**
 - ◆ Gain, filter
- **Calibrations**
 - ◆ Automated through use of additional hardware
- **Measurements**
 - ◆ TC's
 - ◆ Transmitters
 - ◆ Require no excitation



E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Calibration Bus

Test Control Center

GPIB Bus Extender

System Controller (PC)

Tustin CCIS

Signal Conditioning Building

GPIB Bus Extender

Programmable Signal Conditioner

Voltage Standard

Function Generator

Programmable Relay

Programmable Relay

Manual Filter Amplifier

GPIB Cable

Fiber Cable

Multi-conductor Cable

E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Software

- ◆ All of the E-Complex Low Speed DAS software is developed in LabVIEW

- LSDAS Control Software
- Display Screens
- Calibration Software
- Measurement System Analysis (MSA's) Software



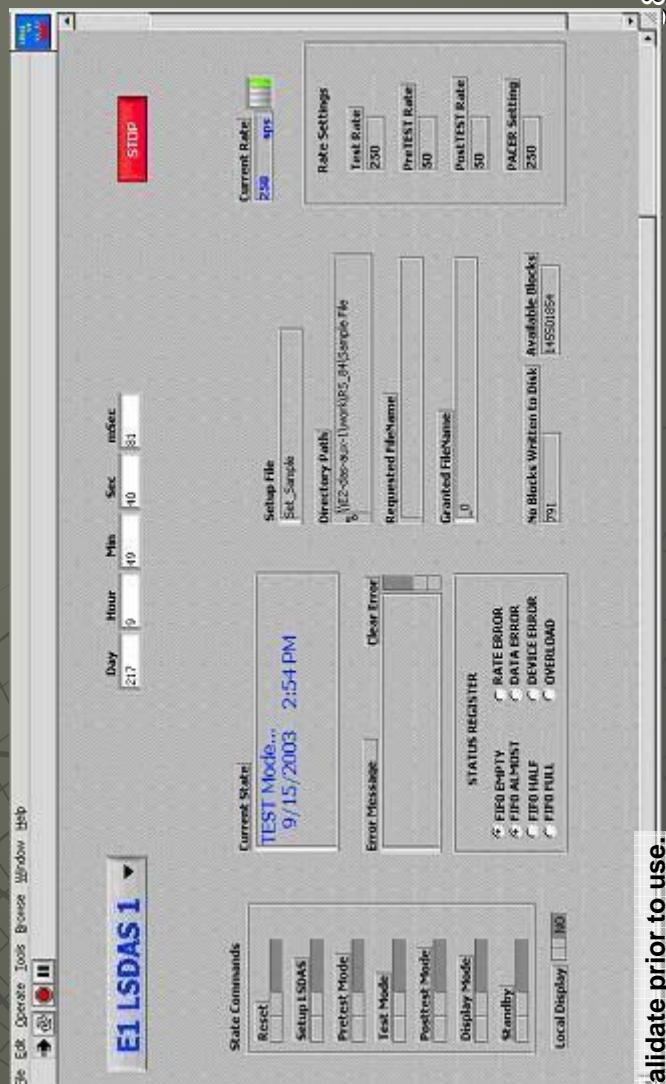
E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Software

- ♦ Low Speed DAS Control Software
 - Used for operation and configuration of the LSDAS Hardware
 - Capability to place system in various modes: Standby, Test, Pre-test, Post-test, Display
 - Saves data to hard-drive
 - Distributes data for remote display



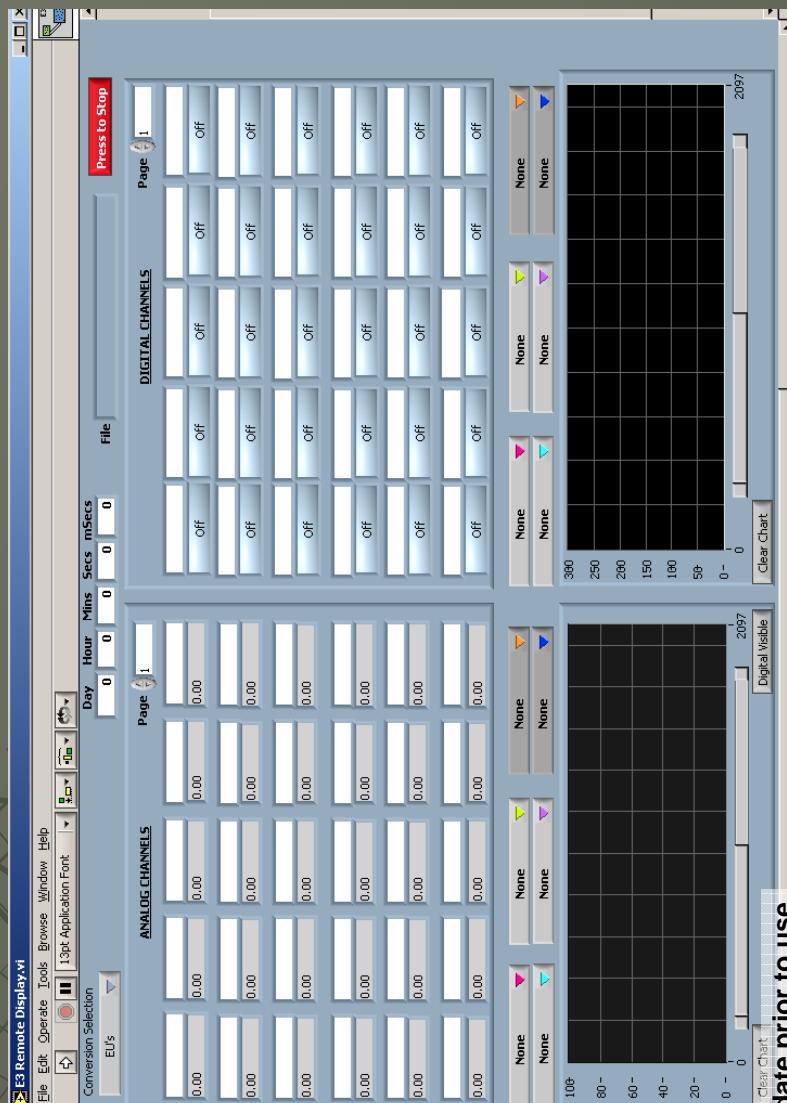
E-Complex

Low Speed Data Acquisition System

Software

◆ Low Speed DAS Display Software

- Tabular and numerical display of measurements
- Analog and digital data



E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Software

Calibration Software

- ◆ Voltage insertion, shunt calibration
- ◆ Calibrate to a tolerance
- ◆ All or subset of channels
- ◆ Generates Report

Additional Functions

- ◆ Setup of programmable amplifiers : gain, filter, excitation
- ◆ Auto-balance
- ◆ Single Channel Diagnostics



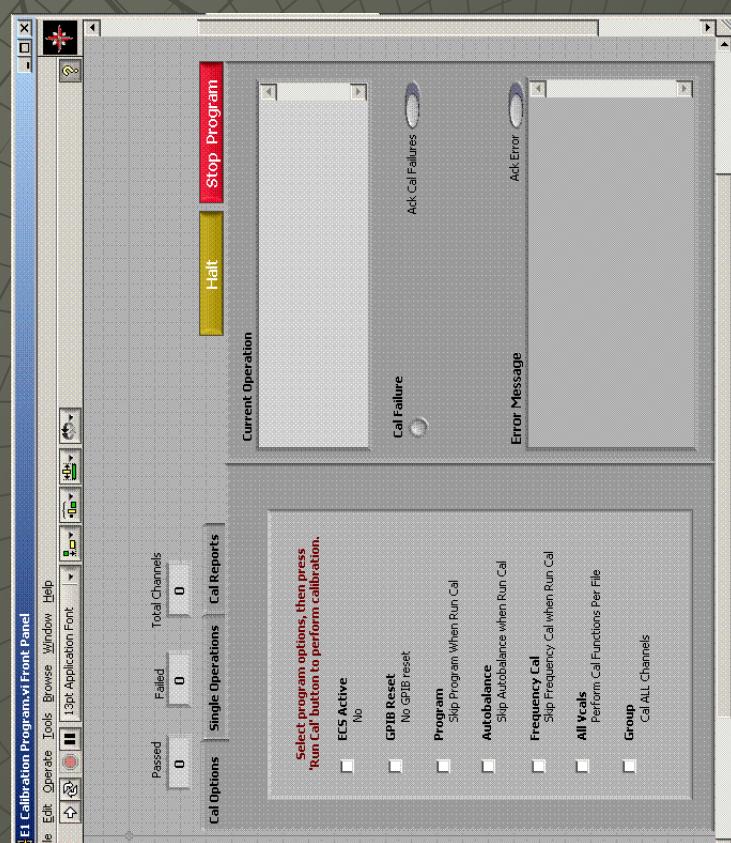
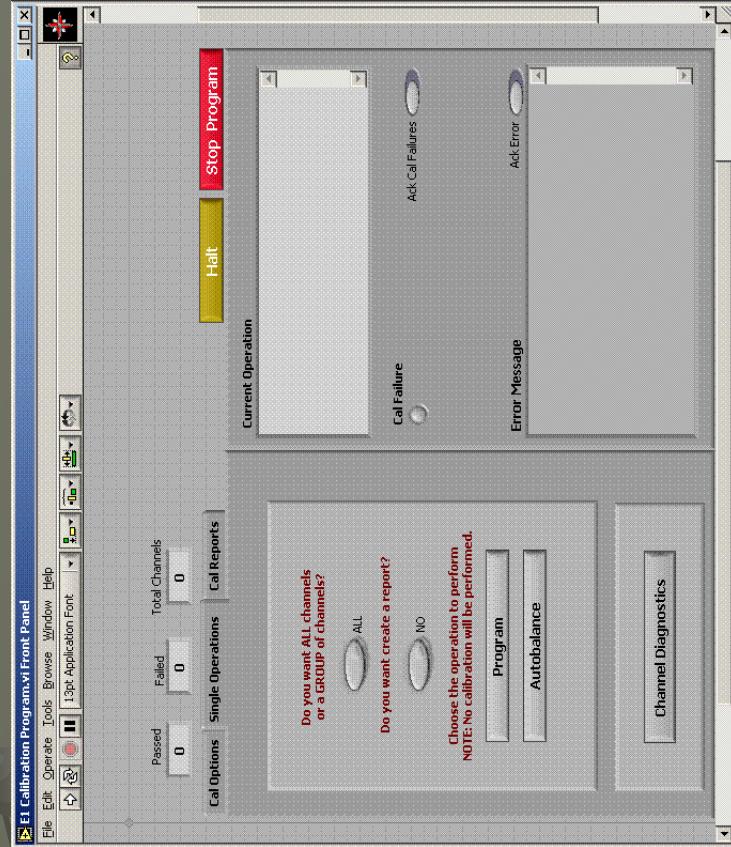


E-Complex

Low Speed Data Acquisition System

Software

◆ Calibration Software

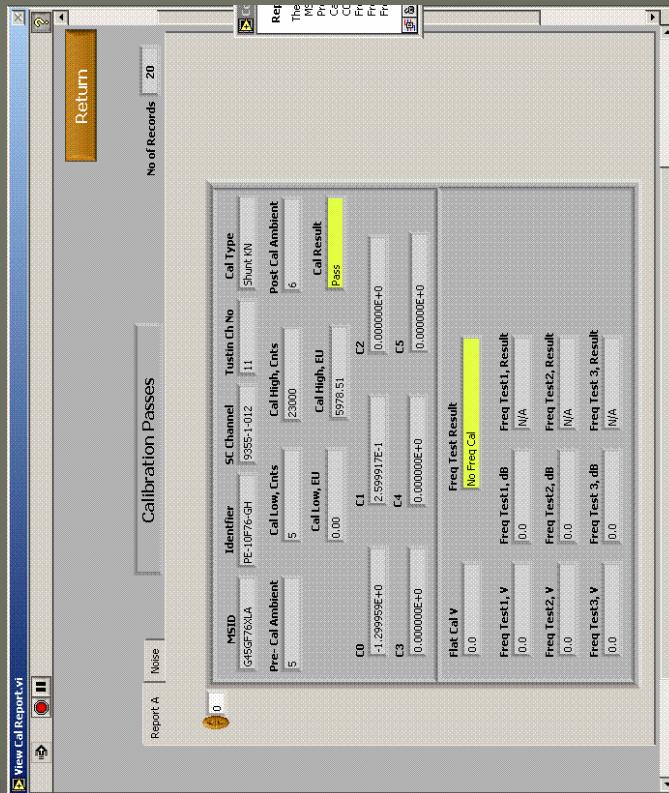
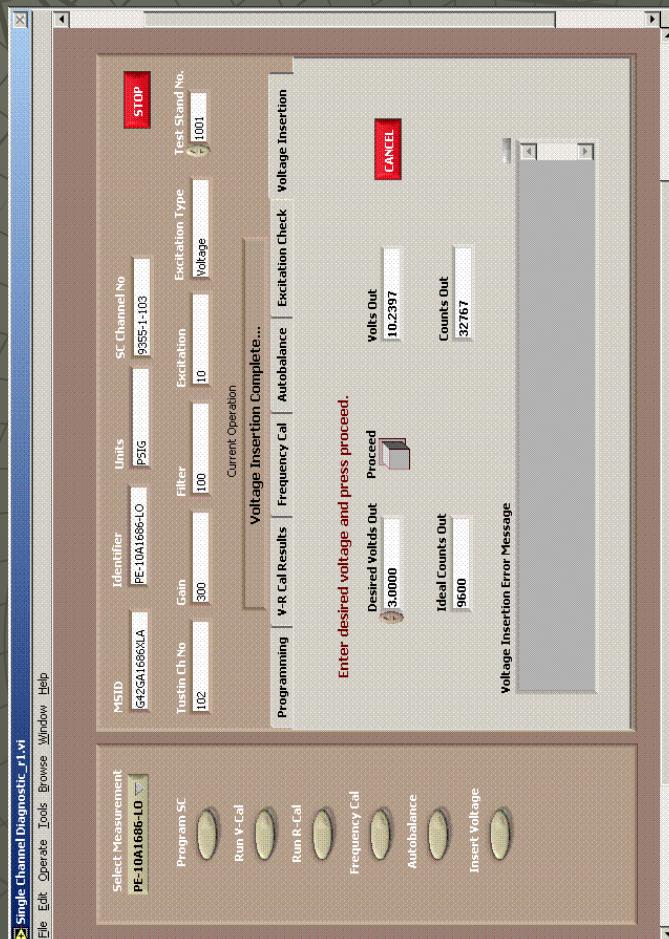


Single Channel Operations

Main Page

E-Complex Low speed Data Acquisition System

Software ◆ Calibration Software



Channel Diagnostics

Report

E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Software

◆ Measurement System Analysis Software

- ◆ Purpose is to quantify a system precision for the LSDAS by evaluating the drift over time of the data system.
- ◆ It consists of a two point calibration performed every hour during an eight hour time span. This is to simulate the maximum time between a pre-test calibration and a test.
- ◆ MSA is performed every thirty days.
- ◆ Reports are generated and data is stored in database.
- ◆ Data from previous runs are used to generate the system's precision and to maintain a history of the data system's response.

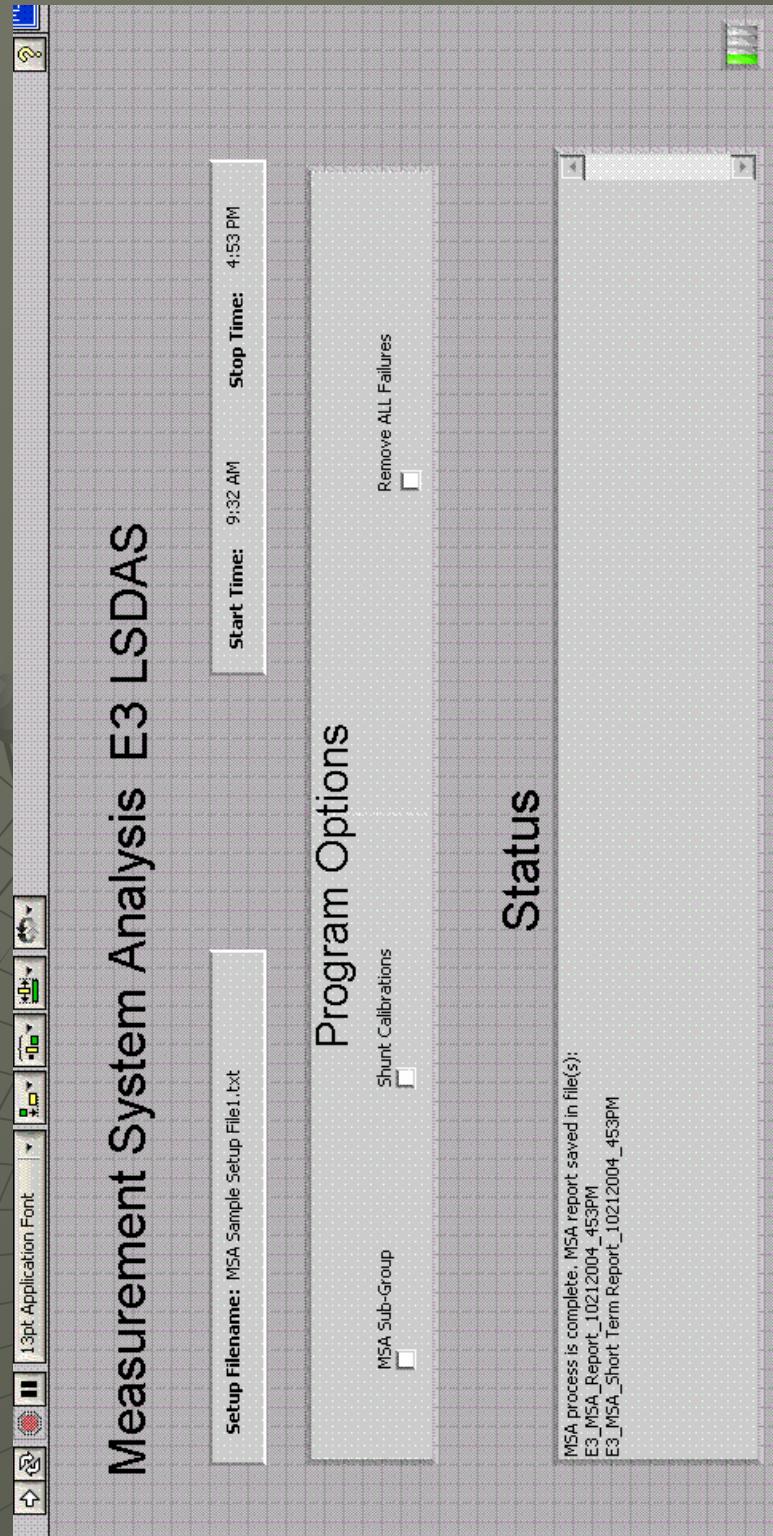
E-Complex

Low Speed Data Acquisition System

Stennis Space Center

Software

◆ Measurement System Analysis Software



Main Screen

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E-Complex

Low Speed Data Acquisition System

Software

◆ Measurement System Analysis Software

Measurement System Analysis ... *in progress*

Hour	1	Complete	Last Cal Time	9:48:21 AM	Next Cal Time	10:48	Taking Data		
9355-1-001	0	6406	6412	25583	25589	9355-1-018	17	6407	6404
9355-1-002	1	6401	6404	25597	25600	9355-1-019	18	6404	6405
9355-1-003	2	6392	6398	25584	25589	9355-1-020	19	6396	6401
9355-1-004	3	6394	6399	25568	25573	9355-1-021	20	6391	6399
9355-1-005	4	6403	6406	25586	25589	9355-1-022	21	6404	6406
9355-1-006	5	6394	6400	25578	25585	9355-1-023	22	6402	6404
9355-1-007	6	6400	6400	25587	25587	9355-1-024	23	6394	6416
9355-1-008	7	6400	6403	25578	25580	9355-1-025	24	6398	6422
9355-1-009	8	6397	6402	25586	25592	9355-1-026	25	6396	6404
9355-1-010	9	6396	6400	25579	25582	9355-1-027	26	6396	6396
9355-1-011	10	6402	6405	25579	25581	9355-1-028	27	6395	6399
9355-1-012	11	6398	6404	25592	25598	9355-1-029	28	6395	6400
9355-1-013	12	6404	6410	25596	25601	9355-1-030	29	6402	6405
9355-1-014	13	6394	6397	25569	25573	9355-1-032	31	6398	6405
9355-1-015	14	6396	6403	25587	25595	9355-1-033	32	6400	6405
9355-1-016	15	6394	6397	25578	25581	9355-1-034	33	6397	6399
9355-1-017	16	6398	6394	25588	25586	9355-1-035	34	6400	6406

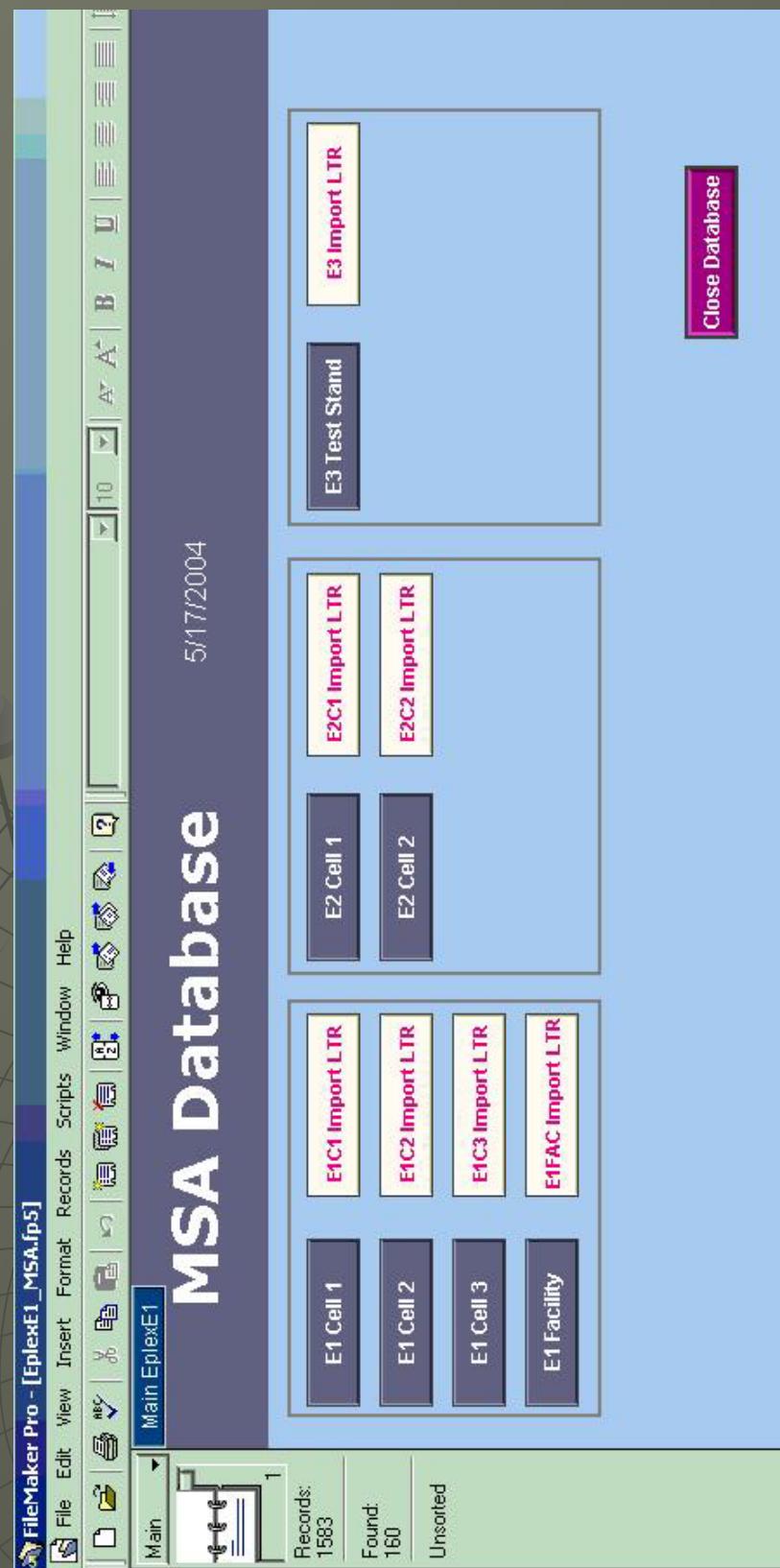


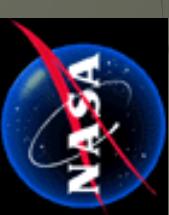
E-Complex

Low Speed Data Acquisition System

Software

Measurement System Analysis Software





AB-Complex Architecture

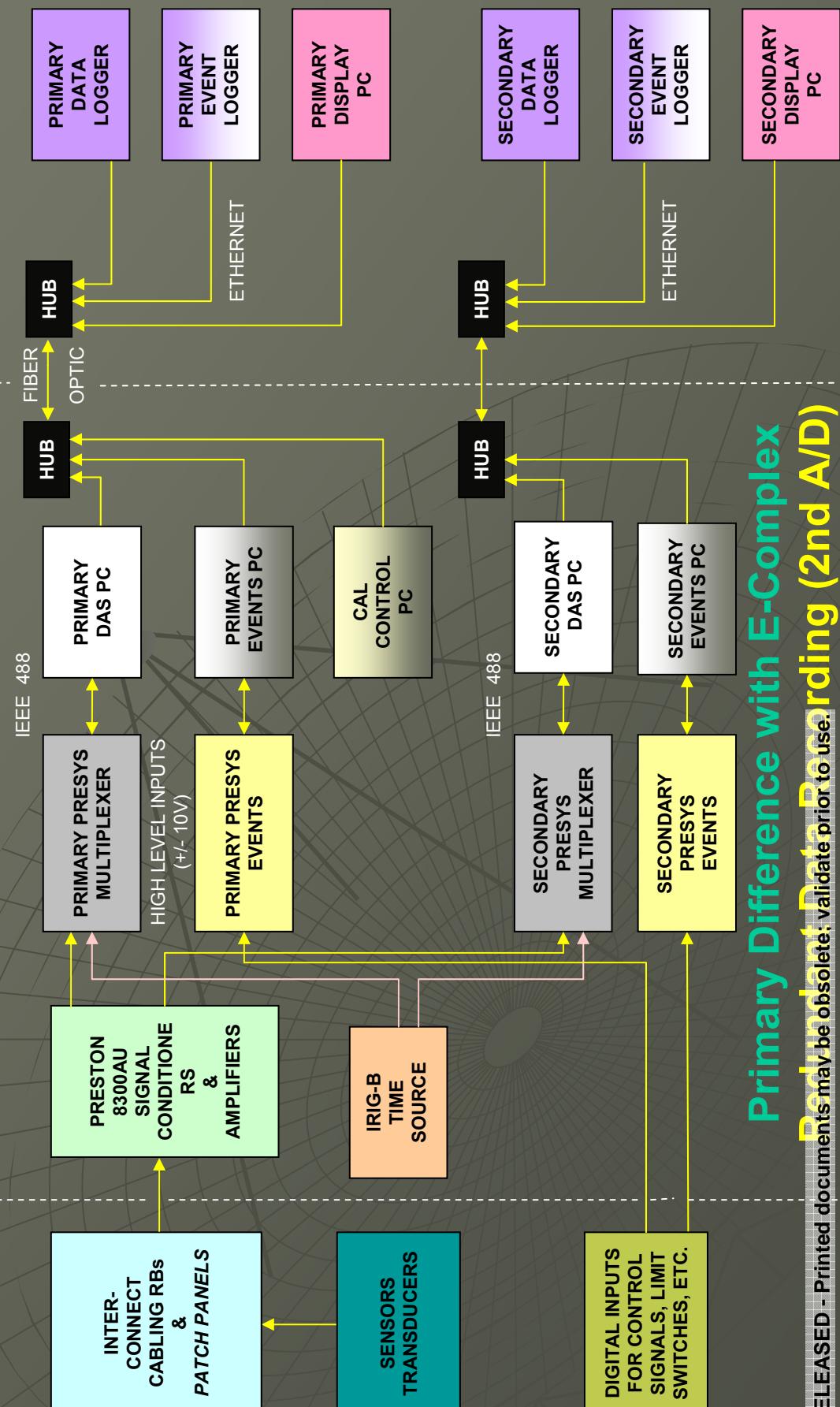
Low Speed Data Acquisition System

Stennis Space Center

TEST CONTROL CENTER

INNER CORE of TEST STAND

OUTER CORE of TEST STAND



Primary Difference with E-Complex Recording (2nd A/D)

It may be obsolete: validate prior to use.

Documents may be obsolete; validate prior to use.

15

AB-Complex Architecture Low Speed Data Acquisition System



Stennis Space Center

◆ The AB-Complex consists of four test stands

- A1, A2, B1, B2 (B1/B2 one structure with two distinct sides)
- ◆ Systems contain 512 analog input channels and 736 digital input channels
- ◆ Each system contains a primary and secondary system for redundancy. Data from the secondary system is only processed if a problem occurs on the primary system.



AB-Complex Architecture

Low Speed Data Acquisition System

Stennis Space Center

PreSys 1000

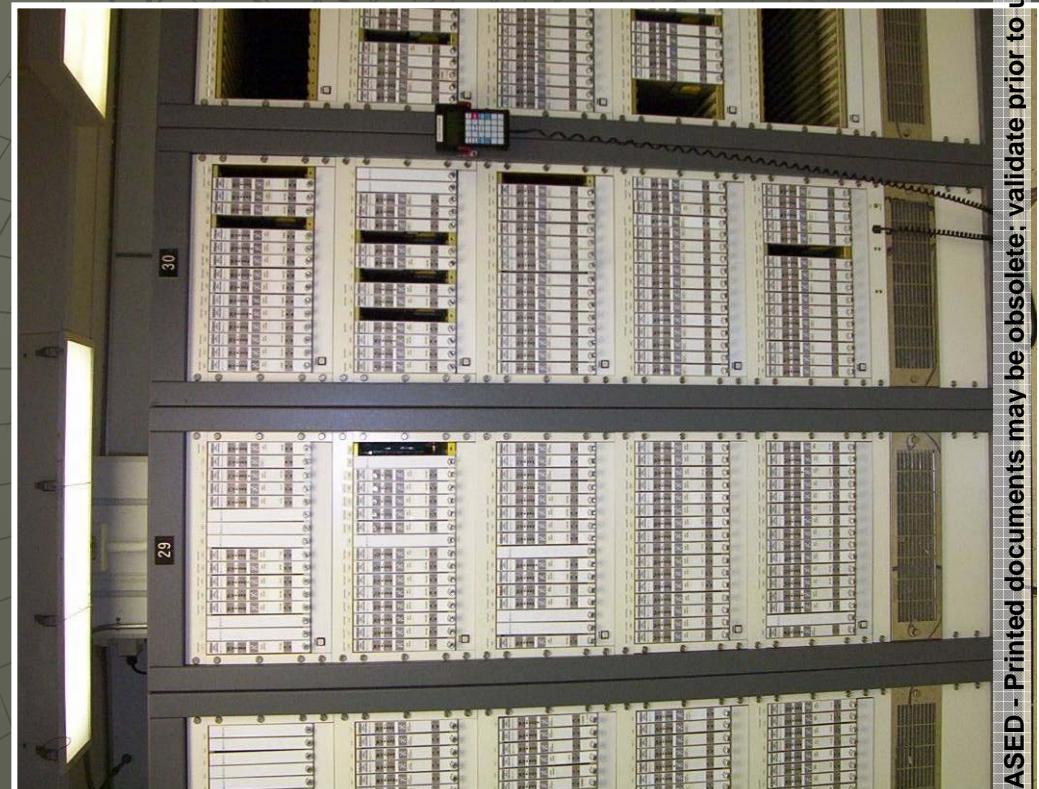
- Fully populated analog box
 - 256 analog input channels
- Fully populated discrete box
 - 480 digital input channels





AB-Complex Architecture Low Speed Data Acquisition System

Preston Amplifiers



Model 8300

- **Programmable**
 - ◆ Gain, filter, excitation
- **Automated calibration**
 - ◆ Voltage Insertion
 - ◆ Shunt
 - ◆ Rcal
- **Various Mode Cards**
 - ◆ Strain Gauge
 - ◆ Full Bridge , Half Bridge
 - ◆ RTD
 - ◆ Thermocouple
- **Measurements**
 - ◆ Strain Gauges
 - ◆ Pressure Transducers
 - ◆ RTD's
 - ◆ Thermocouples



AB-Complex Architecture

Low Speed data acquisition System

Stennis Space Center

Calibration Control

Cal Control PC

Preston 8300
Master
Controller

Voltage
Standard

Function
Generator

Digital
MultiMeter

Analogic
ANDSS5400
Or
Neff
470

Selector
Control Panels

Test Control Center

GPIB Cable

SDAS Cal Box

Test Stand
Select Relay

ENGINE EVENTS
CAL FLAGS

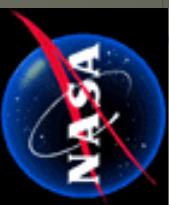
PreSys Events

PreSys MUX

Annadex
Cal Relays

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AB-Complex Architecture Low Speed Data Acquisition System



Software

◆ Software consists of :

- Signal Conditioning Setup
- Measurement Calibration
- Data Acquisition and Real-time Display
- Measurement System Analysis

AB-Complex Architecture

Low Speed Data acquisition System

Stennis Space Center

Software

- ◆ Software written in Microsoft Visual Basic provides computer controlled setups and calibration of the Preston signal conditioners and amplifiers

Signal Conditioning Setup – *Set8300*

- Select gain, filter
- Setup and adjustment of individual signal conditioners and amplifiers

Calibration – *CalMon*

- Automatic calibrations on any number of selected signal conditioners
- Calibrate all active measurements pre-test
- Calibration Types
 - ◆ Shunt Calibration
 - ◆ Voltage Substitution
 - ◆ Excitation Power Supply Calibration

AB-Complex Architecture Low Speed Data Acquisition System

Stennis Space Center

Software

- ◆ **Data Acquisition and Real-time Display – DDAS**
 - ◆ Provides for the control of the data acquisition process and the distribution of data for real-time display
 - ◆ Combines both the analog and discrete data
- ◆ **Measurement System Analysis**
 - ◆ Software originally developed by Rocketdyne
 - ◆ Purpose is to quantify a system precision for the LSDAS by evaluating the drift over time of the data system.
 - ◆ It consists of a two point calibration performed every hour during an eight hour time span. This is to simulate the maximum time between a pre-test calibration and a test.

Typical Low Speed Data Acquisition System Instrumentation



Stennis Space Center

Static Pressure	Static Pressure
Radiometer	Temperature
Temperature	Flow
Flow	LVDT
LVDT	Strain
Level	Proximity
Load Cell	Speed

FACILITY

SPECIAL TEST EQUIPMENT

TEST ARTICLE

• Standard Instrumentation - Not always in the Catalog

- Special Ranges (Cryogenics, Hundreds of Degrees F)
 - Special temperature compensation circuits
 - Special Materials
 - Extremely High Pressures



Typical Low Speed Data Acquisition System Instrumentation

Stennis Space Center

Pressure



Transmitter Delta P

Temperature



Thermocouples

RTD's



Flow

Pressure



Transmitter

Speed



Venturi
Flowmeter



Strain





Control Systems

John Bakker





Control Systems

Stennis Space Center

- The **Control System** manages the test complex and rocket engine or component systems during day-to-day operations and testing while maintaining a safe environment allowing for orderly test shutdown and making facility systems safe in emergency situations.
- Programmable Logic Controllers (PLCs) form the backbone of the SSC Control Systems.
- PLCs primary functions are to sequence rocket engine or component tests and maintain daily operations.
- Hard-wired controls are provided as a backup to the





Control Systems Functions

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◆ Day to Day Operations

- Unloading cryogenics/propellants (Oxygen, Hydrogen, Nitrogen, Methane, etc.)
- Propellant transfers from storage to run tanks
- Pumping up bottle pressures (Nitrogen, hydrogen, helium etc.)
- Gas leak and fire detection.
- Engine drying
- Facility Readiness Test (FRTs)
- Redline cut checks (Redlines are measurements that are monitored by the PLC for the purpose of initiating an immediate shut down when out of tolerance.)





Control Systems Functions

◆ Test Day Operations

- Propellant Transfers
- Engine chill down and prep
- Greenline monitoring (Permissives to start test.)
- Test stand valve sequencing and control during hot fire test
- Redline monitoring during hot fire test
- Performs a controlled shutdown of the engine
 - ◆ Critical valves are also wired to a backup PLC or relays





E1 Test Stand Control System

Stennis Space Center

◆ Three Independent Test Cells

- Can support three different test programs simultaneously
- All test cells share the same propellant run tanks, high pressure bottles, Control System etc.
- Control system must be flexible enough to switch between test cells in twenty four hours

◆ Most Generic PLC (Ladder Logic) of Any Test Facility

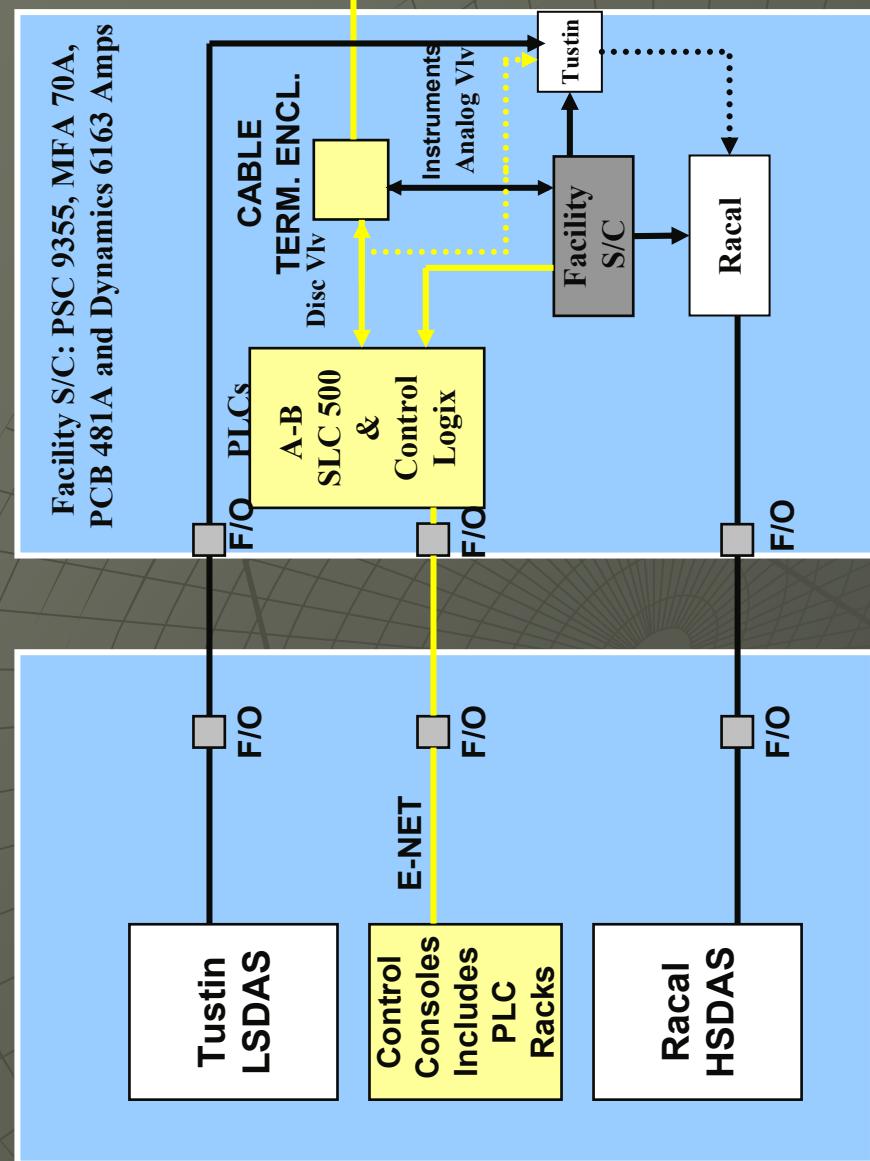
- System is configured entirely through Excel
- Excel tables can be configured in advance and downloaded on test day.
- Excel tables can be archived for historical reference





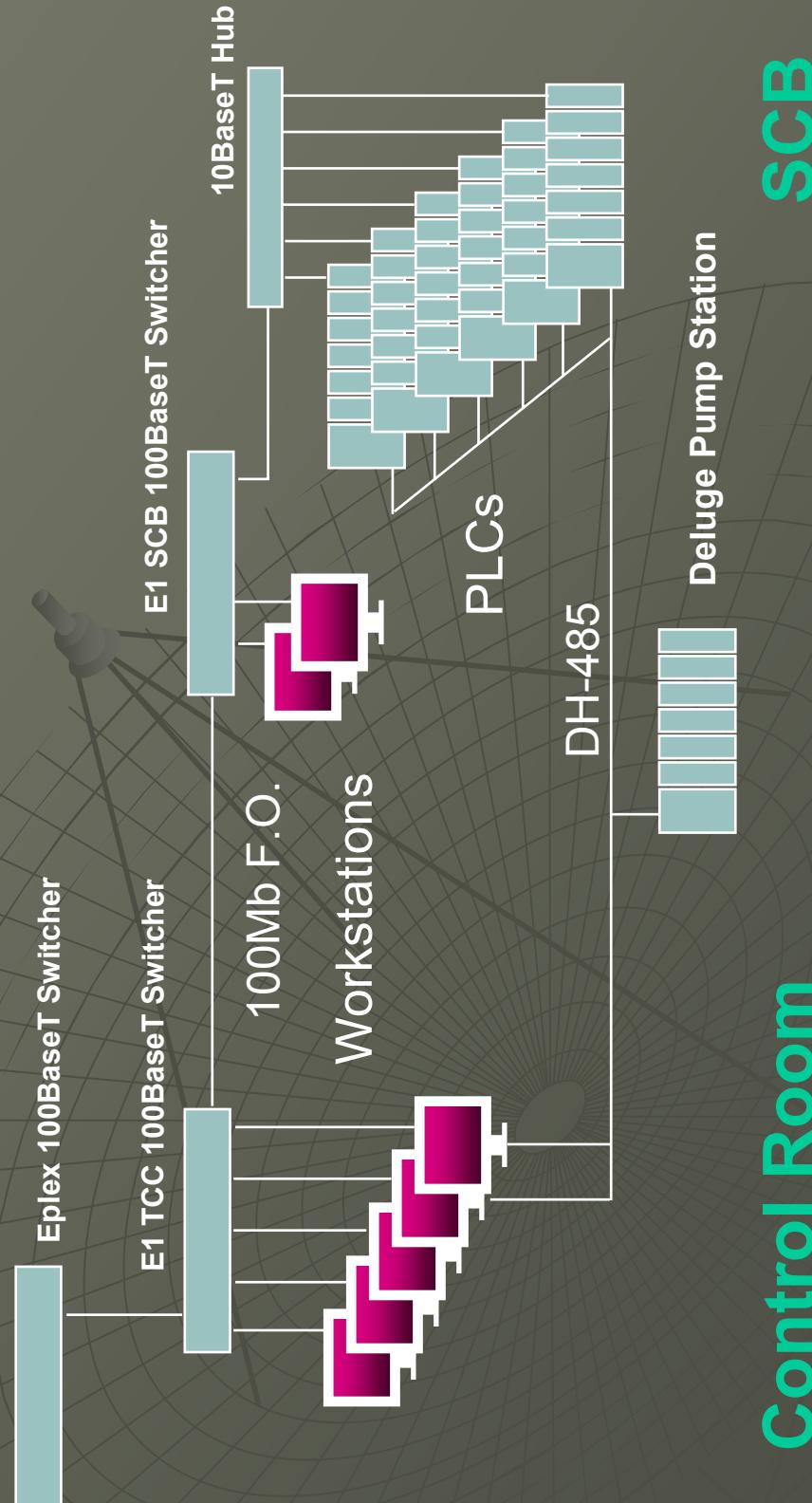
E1 Control System Layout

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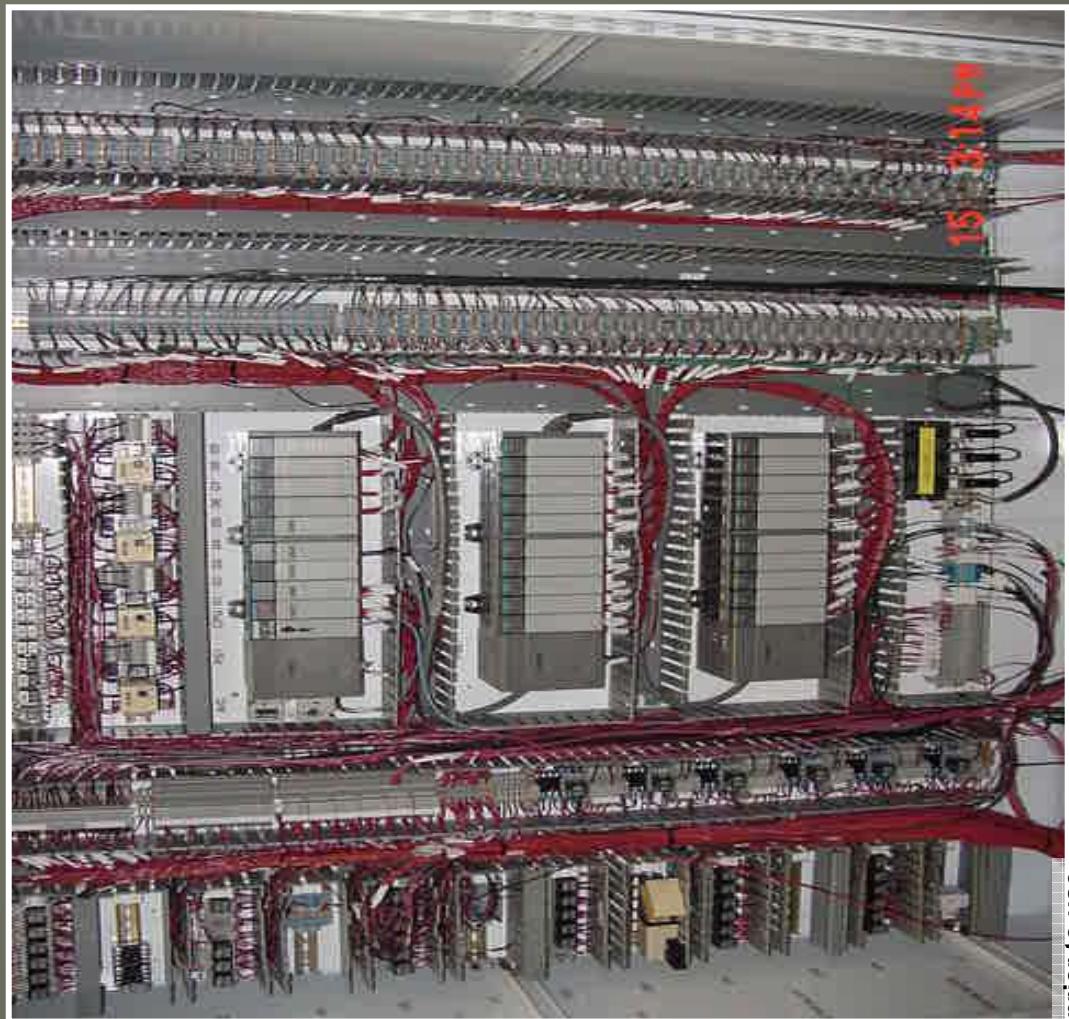
E1 PLC Network Design





Typical E1 SLC Programmable Logic Controller (PLC) Installation

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E1 A-B SLC PLC Cabinet

- Dedicated STE PLC for Cell 2

64 DO	80 AI
12 AO	128 DI

- Shared Display PLC

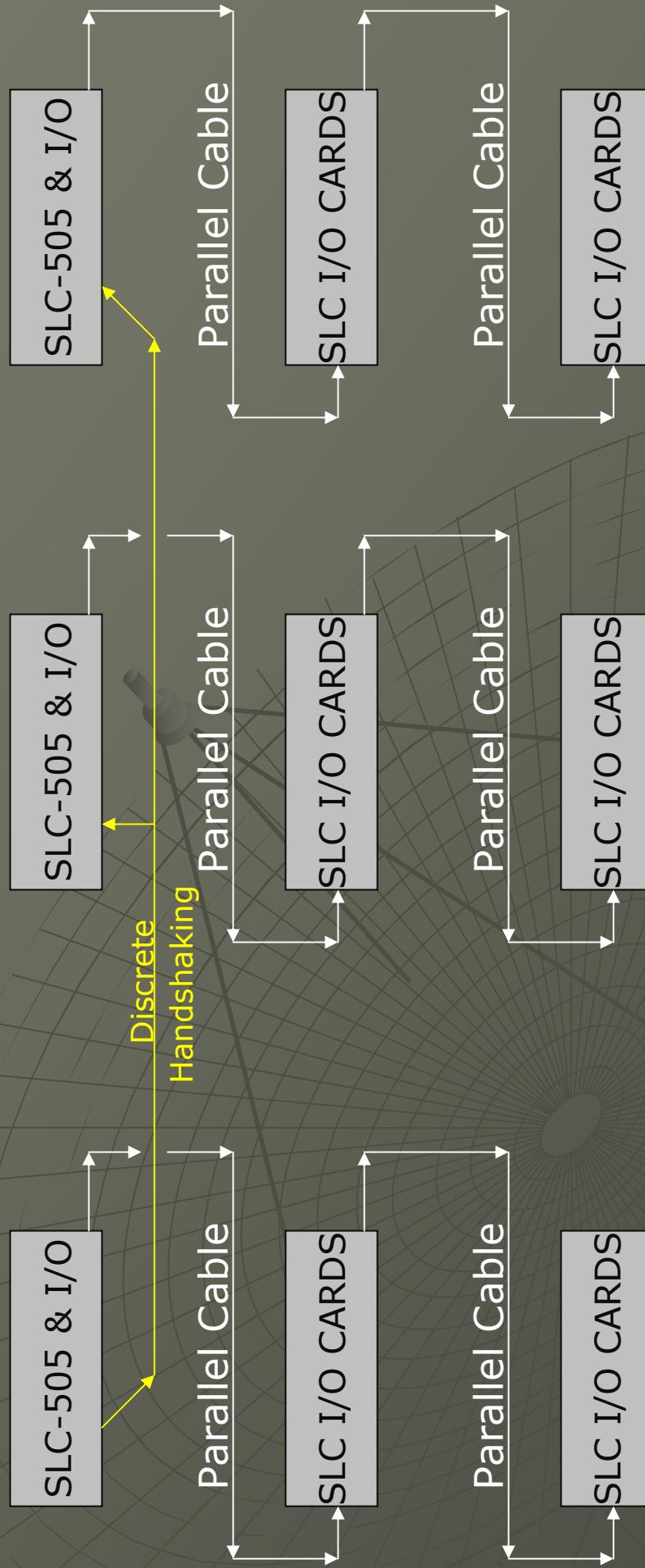
80 AI
32 DI





E1 PLC Architecture with Parallel SLC Input/Output (I/O) Cards

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Advantage: Fast Throughput

Multiple Processors

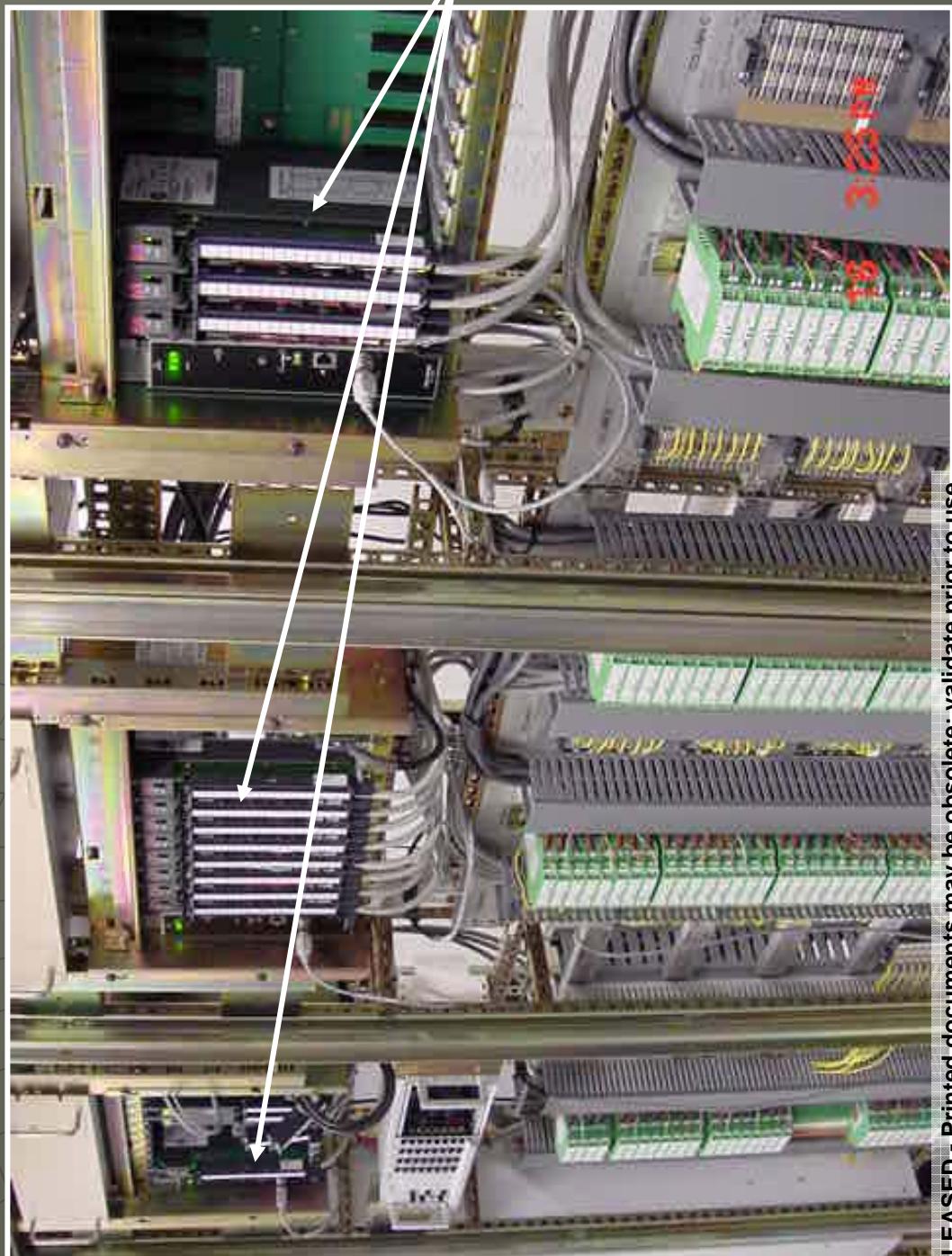
Disadvantages:

Only three racks of I/O per processor

Test Stand E2 Cell 1 Signal Conditioning Bldg 1 Controls I/O



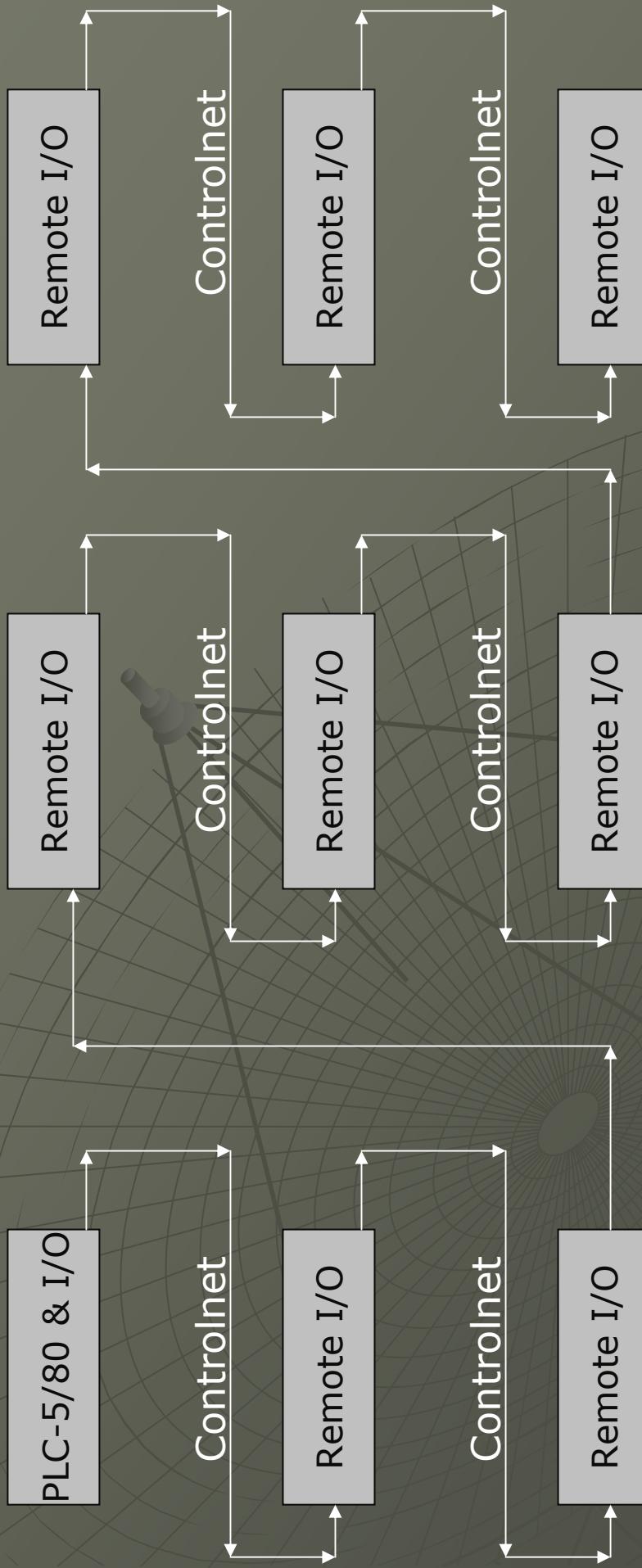
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A/B/E2 PLC Architecture

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Advantages: One Processor

Much larger I/O count

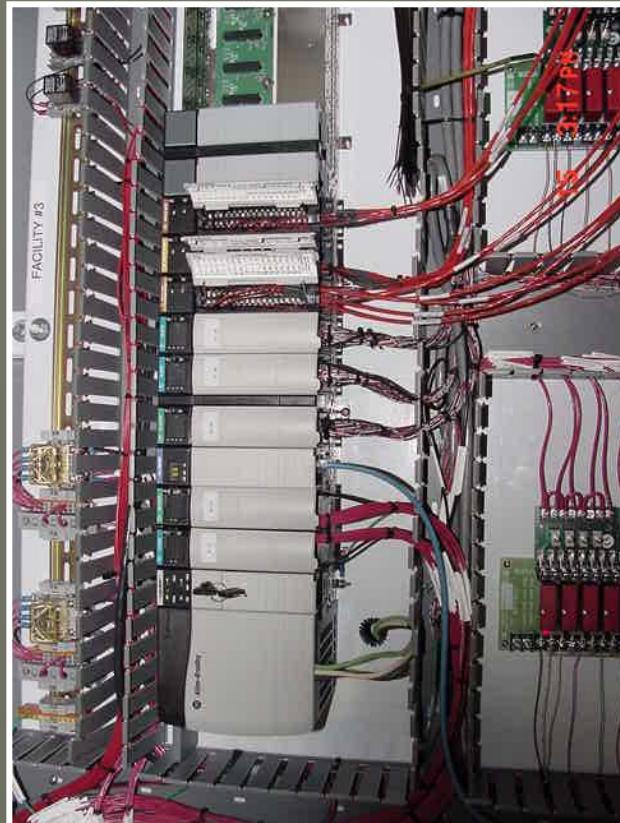
Disadvantages:

Throughput slowed by serial communications



SSC PLC Architecture Changes

Stennis Space Center



Control Logix 5000 PLC

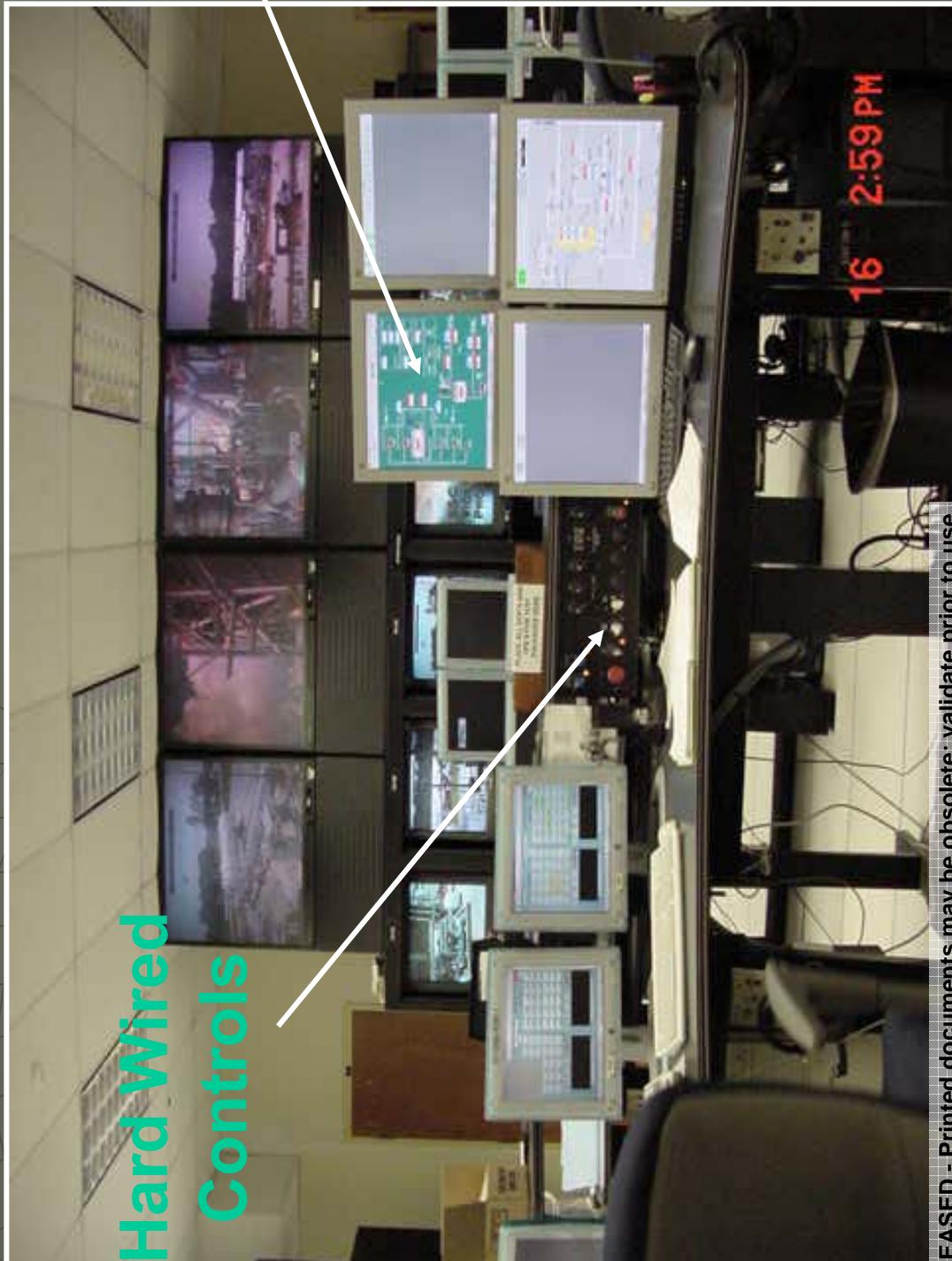
• Migration to faster PLCs in a Distributed Architecture outside the E1 Test Facility

- A-Complex Redline System
- A-Complex Fire & Gas Leak Detect System
- B-Complex Redline System
- B-Complex Fire & Gas Leak Detect System in design
- E3 Redline System

Test Control Center with Human Machine Interface (HMI) Screens



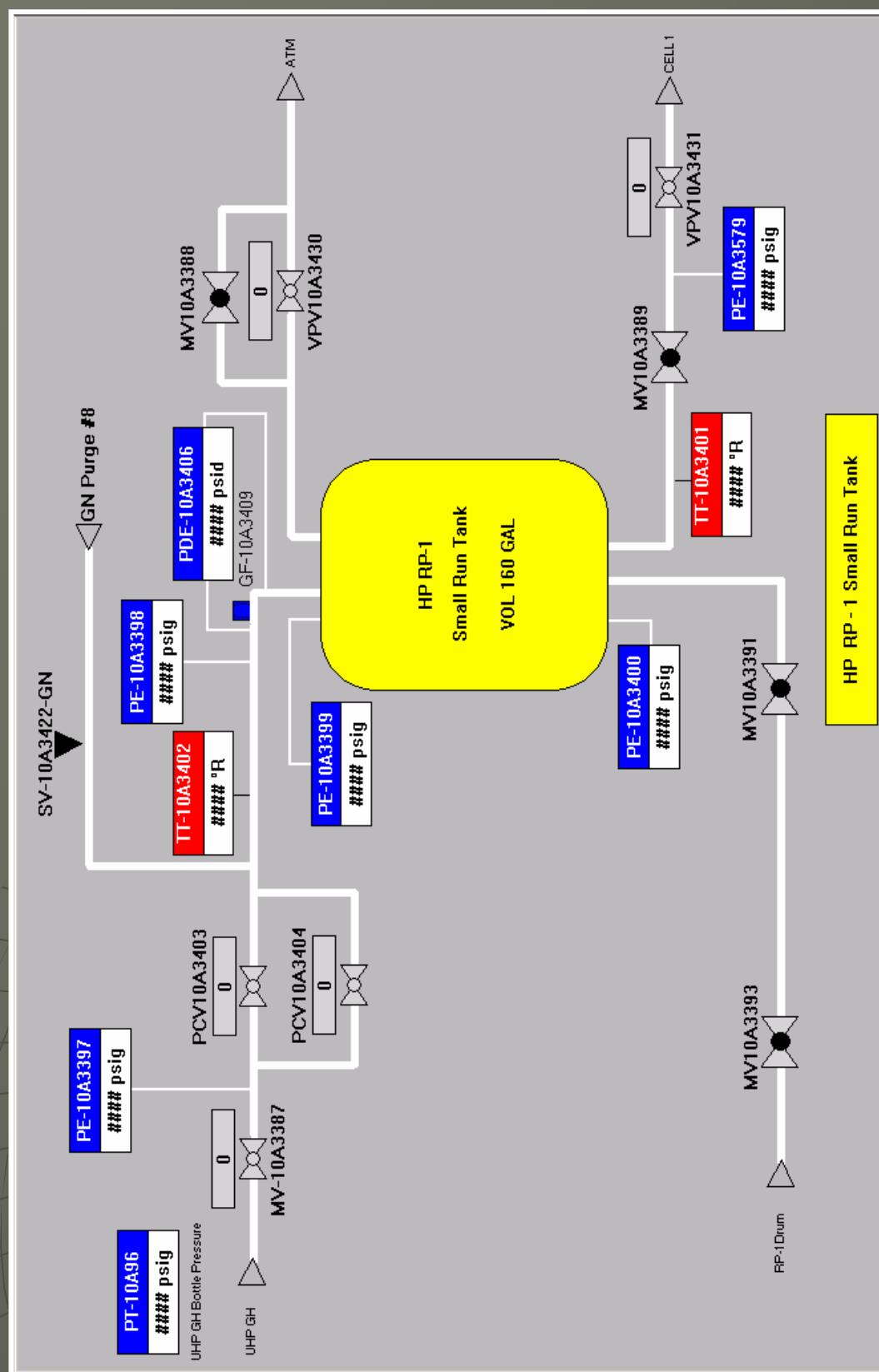
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InTouch by Wonderware HMII

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Data Acquisition & Control Systems Lab (DACS Lab)

Scott Jensen





DACS Lab

Stennis Space Center

- The **DACS Lab** is a facility designed to provide an off-stand capability for developing data acquisition and control systems in support of testing.

- Safe and controlled environment allows verification and development without impacting project schedules or compromising pre-existing test hardware, software, networks or configurations.
- Useful in the identification and resolution of significant issues with equipment and configuration functionality prior to activation.
- Servo valve control capability and personnel's expertise have been utilized to expedite mission critical valve integrity checks prior to field installation.
- Helps to eliminate facility downtime and test delays.
- Provides for hands-on training, qualifying spares, market evaluations, minor equipment repairs, and familiarization with data acquisition and controls equipment



DACS Lab

Stennis Space Center



Controls Station

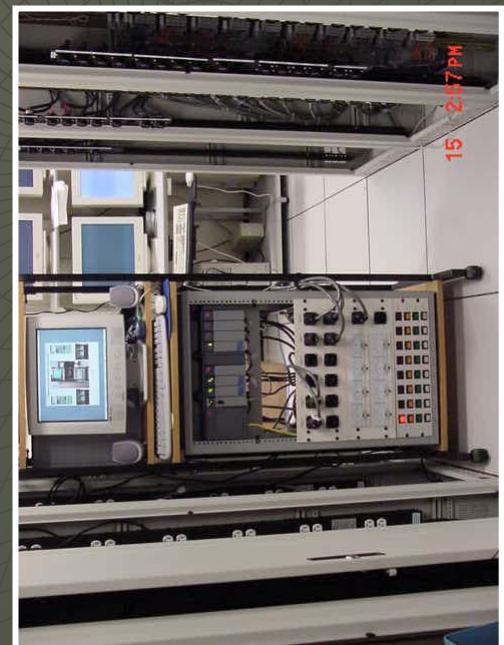


Sensor Workstation

RELEASED - Printed documents may be obsolete: validate prior to use.



DAS Station



Controls Checkout Cart



Sensor Development





Wireless Sensor Development

Stennis Space Center

Sensors Needed to Monitor Valve Health

High-Geared Ball Valves

➤ Torsional shaft strain

➤ Total valve cycles

➤ Cryogenic valve cycles

➤ Inlet temperature

➤ Outlet temperature

➤ Body temperature

Linearly Actuated Valves

➤ Linear bonnet strain

➤ Total linear travel

➤ Total directional changes

➤ Valve preload position

➤ Inlet temperature

➤ Outlet temperature

➤ Body temperature



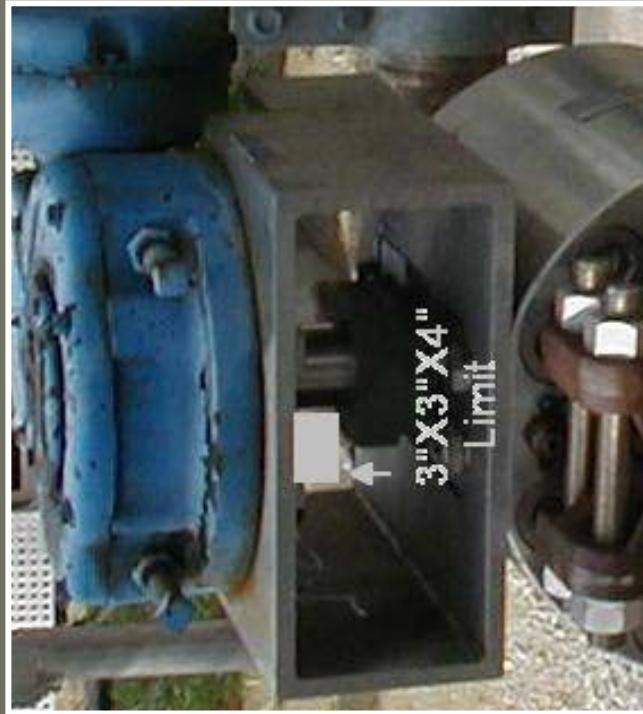


Wireless Sensor Development

Stennis Space Center

Confined Locations

- Sensor Size
 $2\frac{1}{2} \times 3 \times 4$ inch
- Wireless
radius
35 foot transmission
- Battery powered
Two battery packs
with two 9 Volts
supplies





Wireless Sensor Development

◆ NEC Class I Division II B Hazardous Environment

- Compliance with NEC article 501

- Enclosed in Potting

Blue-epoxy flame retardant 832FRB

M.G. Chemicals

- Internal temperature monitoring

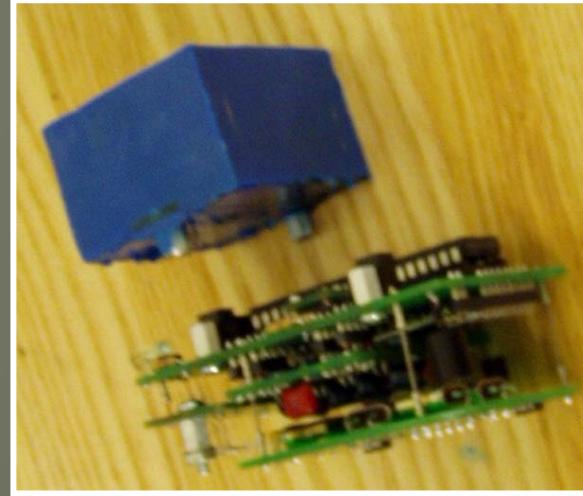
Shutdown 150° F

- No exposed arcing points

- Limited operational power

9 Volts at 250 milliamp

- No exposed cavities





Wireless Sensor Development

Stennis Space Center

Power Conservative

- One-Way Communications

Linx HP3 transmitter and receiver modules

- Microprocessor Sleep Mode

- Piezoelectric Wake-Up Circuitry

Measurement Specialties LDT series

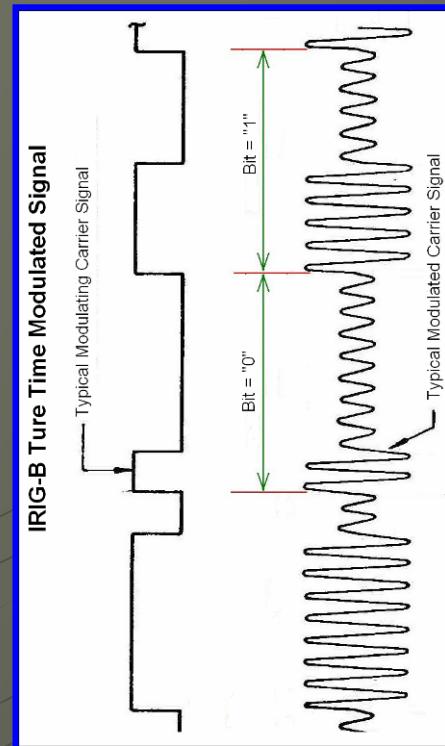
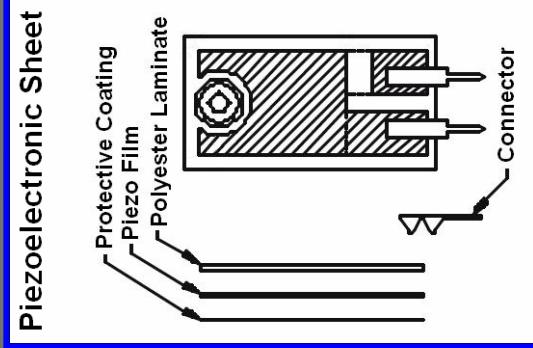
Accurate Data Synchronization

- IRIG-B Timing Module

Facility correlation

- Communication Bus

Internal data correlation



Wireless Sensor Development

◆ Automatic and Manual Data Access

- **Memory storage Network capable**

Compact flash card memory access
ARMA Design Inc.

- **Network capable**

Ethernet broadcast I-7188E
ICS DataCom Inc.

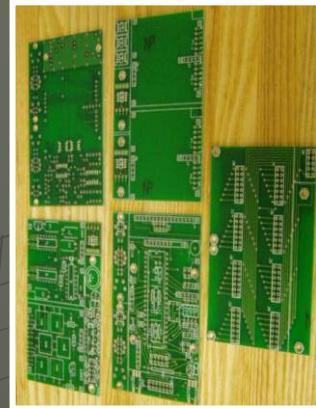
◆ Setup and Maintenance

- **Simple Human interface**

Switch and Indicator light

• On-board programmer interface

Serial communications
Software updates



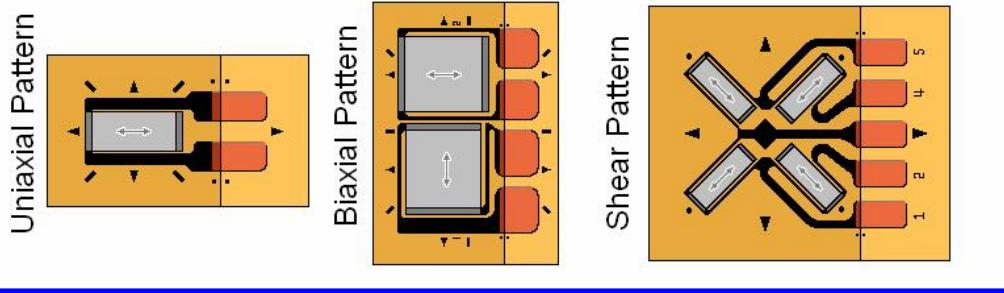


Wireless Sensor Development

◆ The K-Type thermocouples sensor

(for inlet, outlet, and body temperature monitoring)

- Monolithic thermocouple amplifier from Analog Devices
- Uses cold junction compensation



◆ The strain instrumentation sensor

(for bonnet and torsional strain monitoring)

- Axial Strain by a Vishay precision quarter bridge
- Biaxial Strain by a Vishay precision half bridge
- Shear Strain by a Vishay precision full bridge



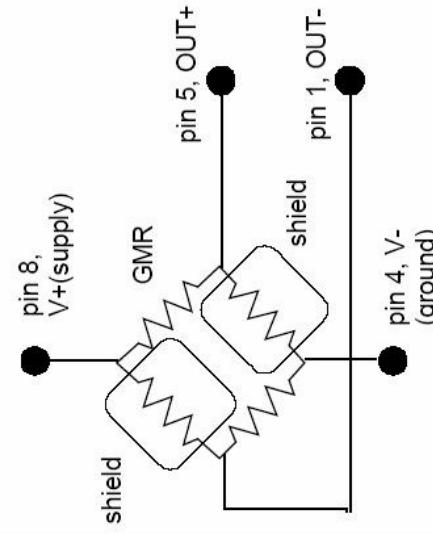
Wireless Sensor Development

• **Limit switch sensor**

(for monitoring number of cycle)

- **6 magnetic reed switches**
- **4 input with wake-up abilities**

• **Giant Magneto Resistive**



• **Signal interface sensor**

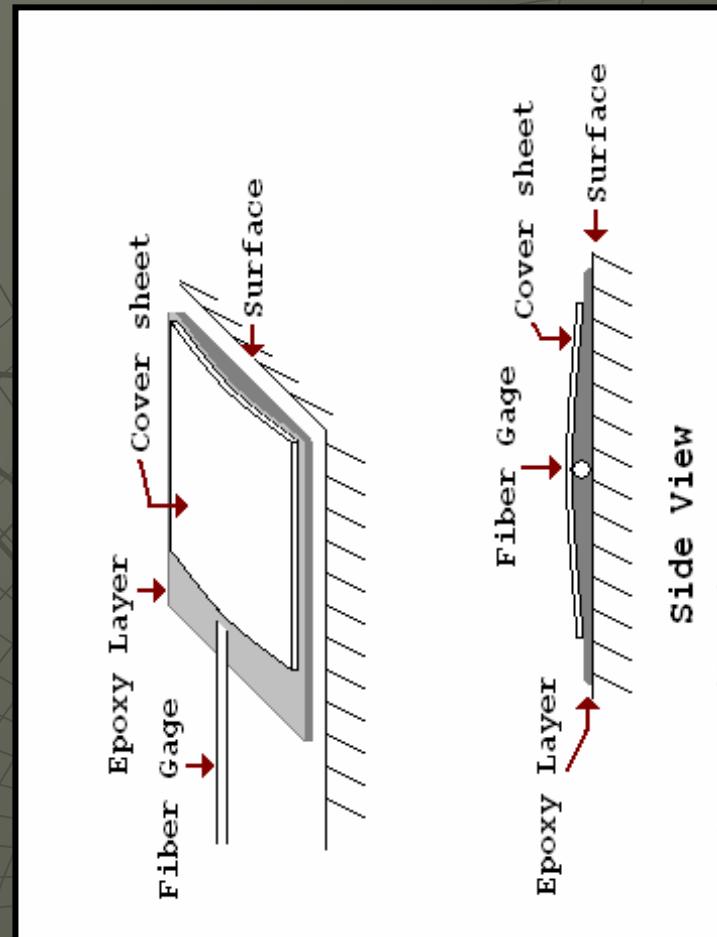
(for Linear Voltage Differential Transformer (LVDT) monitoring)

- **4 to 20 milliamp current loop signal**
- **Giant Magneto Resistive (GMR) from Unobtrusively monitors magnetic fields**
- **0 to 10 volts Direct Current (DC) signal**
- **Basic voltage follower circuit**



Cryogenic Sensor Development

- ◆ Developed improved bonding techniques for strain gauges and thermocouples used in cryogenic service



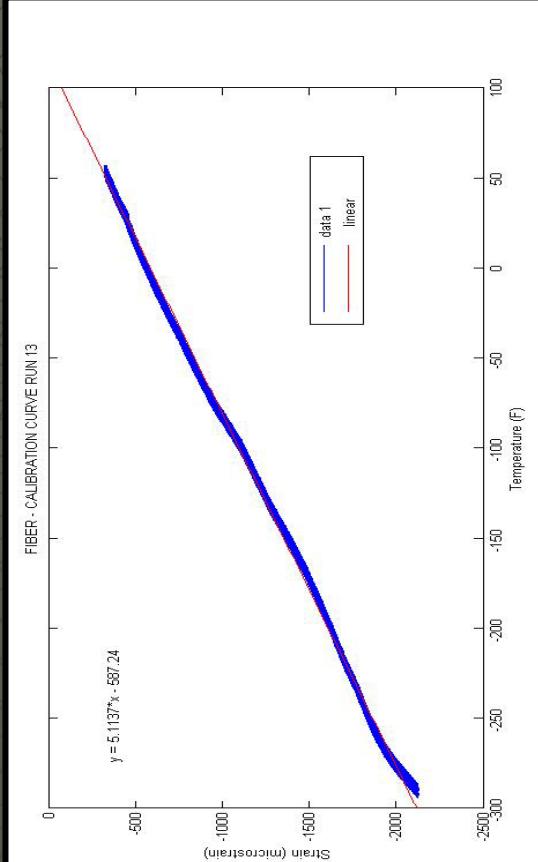


Cryogenic Sensor Development

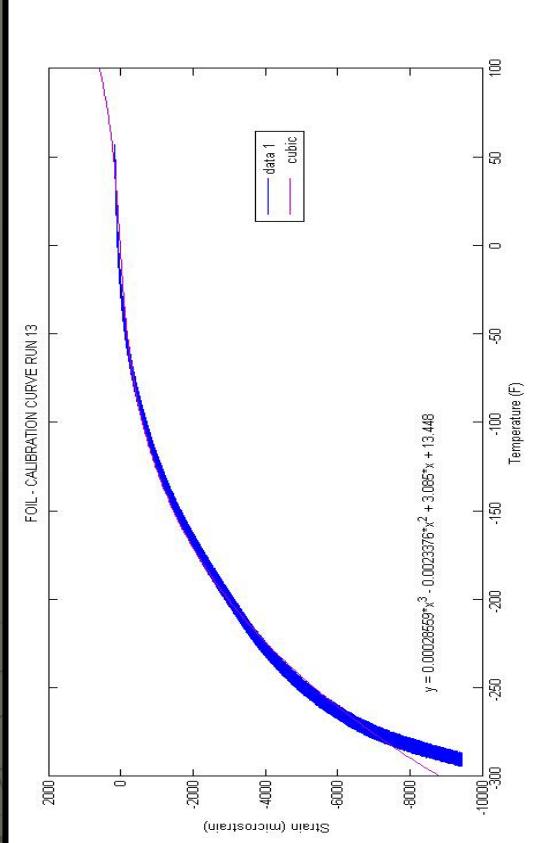
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- ◆ Developed calibration curves for foil and fiber optic strain gauges at cryogenic temperatures

Fiber Optic Strain Gauge Curve



Foil Strain Gauge Curve



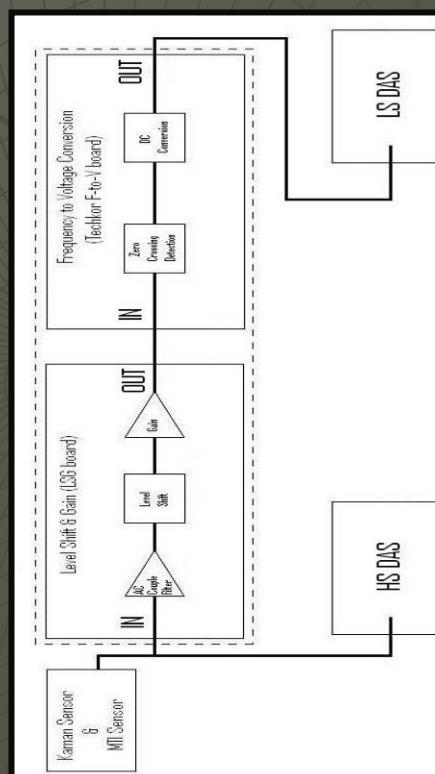
Speed Sensor Signal Conditioning Development



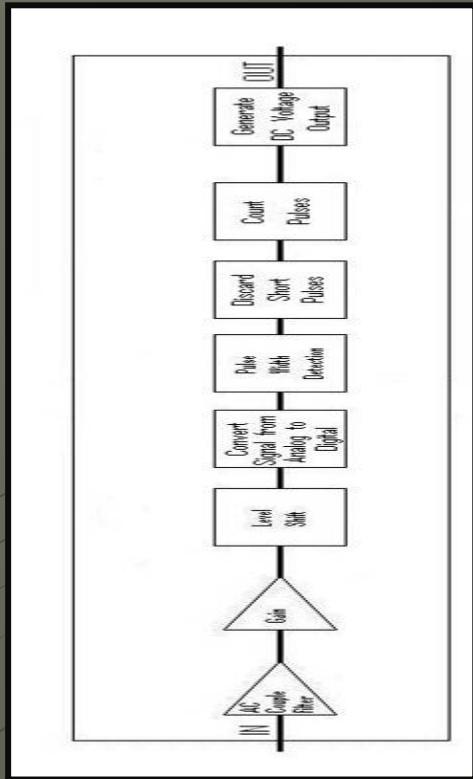
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- ◆ Developed a frequency to voltage converter for determining rotational speed of turbopumps during rocket engine testing - improved the response to complex waveforms recorded from speed sensors

Existing Speed Detection Logic



Experimental Speed Detection Logic

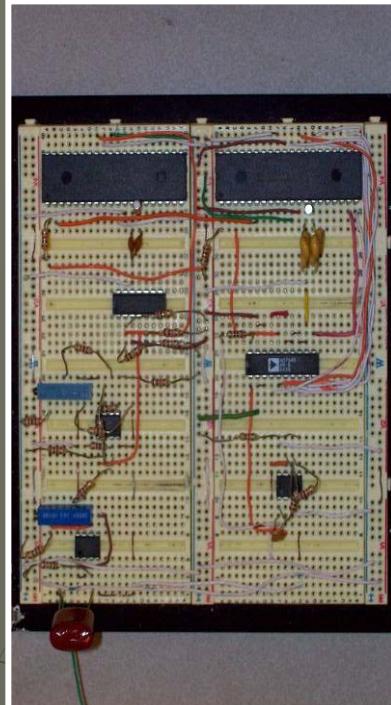


Speed Sensor Signal Conditioning Development

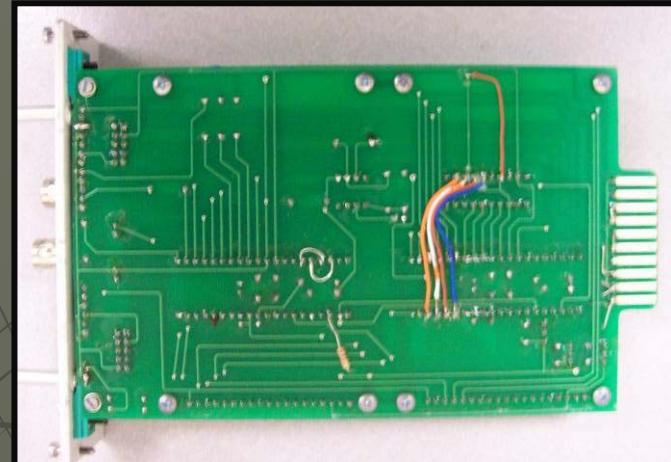


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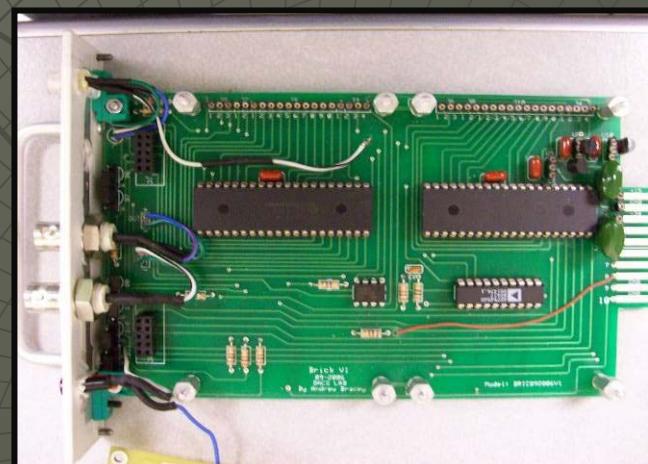
Breadboard



Populated Board (bottom)



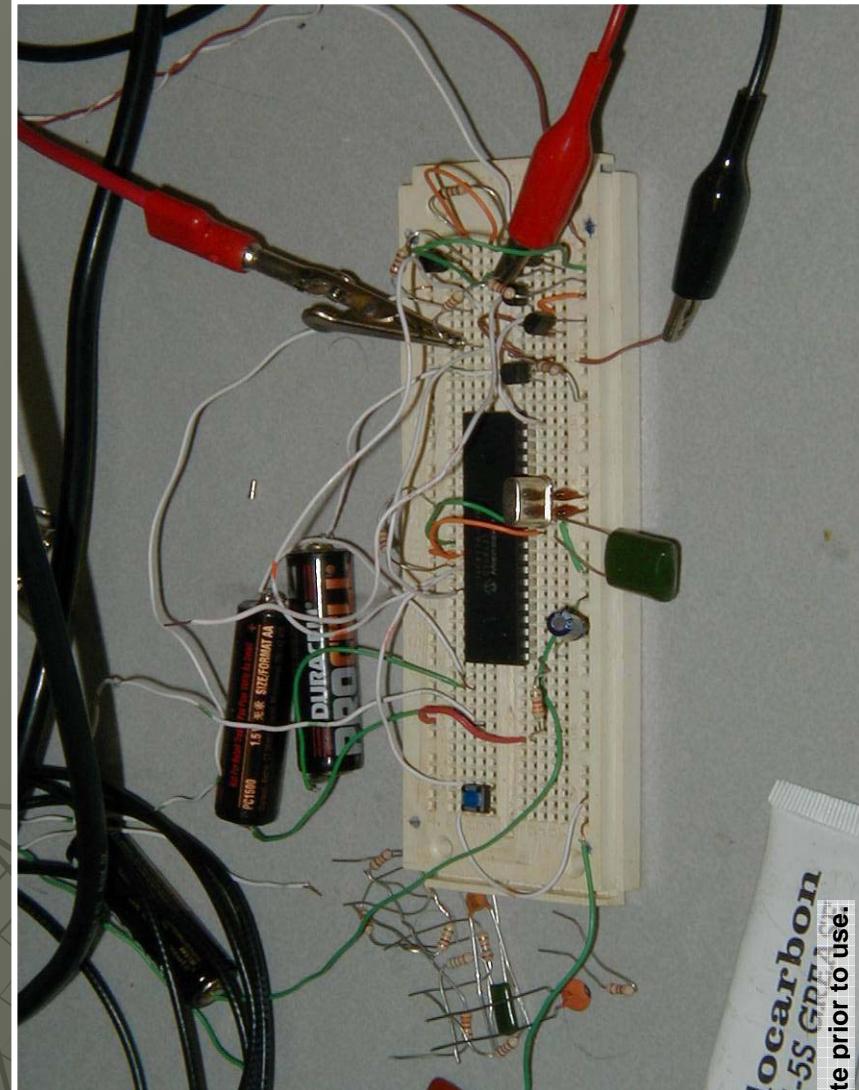
Populated Board (top)



Piezoelectric Sensor Health



- ◆ Developing the techniques to evaluate the health of piezoelectric sensors



Breadboard of
Piezoelectric
Sensor Tester



Summary

- ◆ NASA/SSC's Mission in Rocket Propulsion Testing Is to Acquire Test Performance Data for Verification, Validation and Qualification of Propulsion Systems Hardware

- Accurate
- Reliable
- Comprehensive
- Timely

- ◆ Data Acquisition in a Rocket Propulsion Test Environment Is Challenging

- Severe Temporal Transient Dynamic Environments
- Large Thermal Gradients
- Vacuum to 15 ksi pressure regimes

- ◆ SSC Has Developed and Employs DAS, Control Systems and Robust Instrumentation that Effectively Satisfies these Challenges